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<b>SUMMARY</b>				
<p>The effect of the free surface is of great interest for marine technology, however including the free surface in computational fluid dynamic (CFD) predictions complicates the problem. This study examines the processes needed to include the free surface in CFD predictions for a simplified ship hull at yaw angles up to 40 degrees. The CFD predictions were compared to experiment data whenever possible and to predictions for the same flow conditions in a fluid with no free surface. The results are compared using calculated force coefficients, flow patterns and where applicable, free surface wave profiles.</p>				
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océaniques

## **EFFECT OF FREE SURFACE ON CFD PREDICTIONS FOR WIGLEY HULL FORM AT INCREASING YAW ANGLES**

SR-2006-20

Amanda Collier, David Molyneux

August 2006

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# **EFFECT OF FREE SURFACE ON CFD PREDICTIONS FOR WIGLEY HULL FORM AT INCREASING YAW ANGLES**

## **INTRODUCTION**

Computational fluid dynamics (CFD) has become an integral part of designing ship hulls and analyzing hydrodynamics. The computations allow the hull designer to predict the wave patterns around a ship hull and the forces and pressures caused by the flow. Most recently it has become possible to include the effect of the free surface on these parameters within commercial CFD codes. Traditionally, the free surface effects were often ignored due to the complexity of the problem. Increased computer capacity and improved numerical schemes have made the calculation of free surface viscous flows a realistic goal for CFD studies.

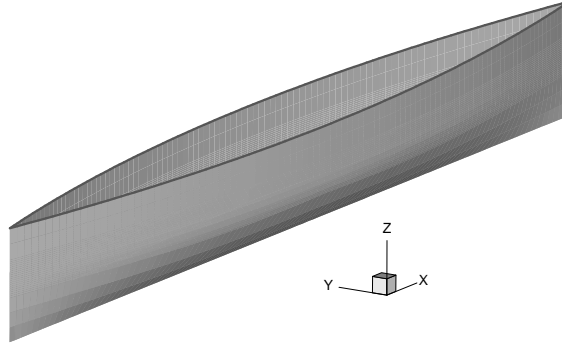
In marine cases, the free surface is especially important because ships operate at the interface between two fluids. However, including the free surface is challenging, making the design of the mesh and the definition of the problem more complex and increasing the number of iterations for the CFD program to reach a solution. The work described in this report is an attempt to determine how significant the free surface is to the computed forces and flow patterns for a ship hull with a yaw angle, given that in most practical situations a ship cannot sustain a high yaw angle at Froude numbers over 0.2, and below this speed, the waves generated are generally small.

The chosen hull shape for this study was Wigley hull, defined by second order curves of  $x$ ,  $y$  and  $z$ . It is a very simple shape, simplifying the mesh and the time required to design it. Despite its geometric simplicity, this hull has been well studied by other researchers, with data from model experiments and CFD predictions available.

For this study, the effect of free surface flow on wave patterns, velocity vectors and forces at the hull will be examined at several different yaw angles. The study described in this report was carried out using the commercial RANS based CFD program *Fluent* (Fluent Inc., 2005) and meshes for this program were created using *Gambit* (Fluent Inc., 2005), the meshing program supplied by the same company. To study the free surface effects, a mesh with fine definition around the free surface was designed, and CFD predictions for several different yaw angles were obtained. These results were compared to the results of a single-phase system at the same flow speeds, using the same mesh but with the cells above the free surface removed. Both sets of results were compared to experimental results and/or published computational results wherever possible.

## **MESHING STRATEGY**

Before meshing could begin, a file defining the Wigley hull surface [1] was imported into *Gambit*, with the surfaces being imported as virtual surfaces. The hull shape is shown in Figure 1. In order to obtain acceptable results (for example, good definition at the free



**Figure 1:** Wigley hull form

surface and around the hull), several meshes were developed, with each mesh improving on the deficiencies of the previous mesh. This process was somewhat trial and error, and only the final mesh used to carry out the study will be described here.

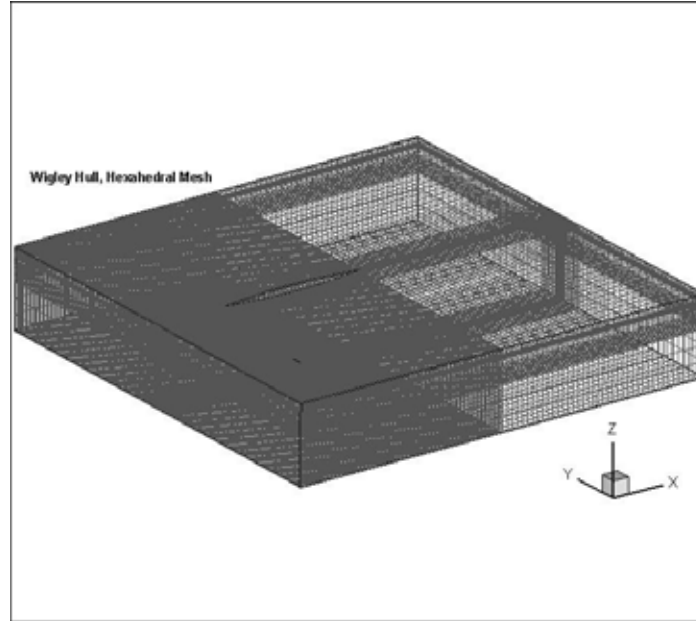
The origin was oriented so that the  $x$ -axis was positive towards the stern, the  $y$ -axis was positive to starboard and the  $z$ -axis was positive upwards. The origin itself lies at the center of the hull. Hexahedral meshes were used for the entire domain, thus all faces had to be four-sided, and all volumes made up of six faces. As the main objective of the proposed numerical experiments was to evaluate the effect of the free surface, it was particularly important to create a very fine grid in the area where the free surface was expected to exist.

The free surface is initially assumed (by *Fluent*) to be at  $z = 0$  (the waterline of the hull when stationary in calm water), thus the volumes around this plane had to be meshed with extremely small cells. Beginning at this level, a very small volume was created just above the free surface, and just below. The surface of the imported hull was split into eight faces (four on each side), thus each side of the hull had four volumes along the  $x$ -direction. These faces were further split to create the fine mesh at  $z=0$ . In the end, 32 faces were used to define the hull. Once the volumes immediately above and below the free surface were created and meshed with a very fine grid, the mesh was continuously built out from the hull, with the meshing becoming coarser as the distance from the hull increased.

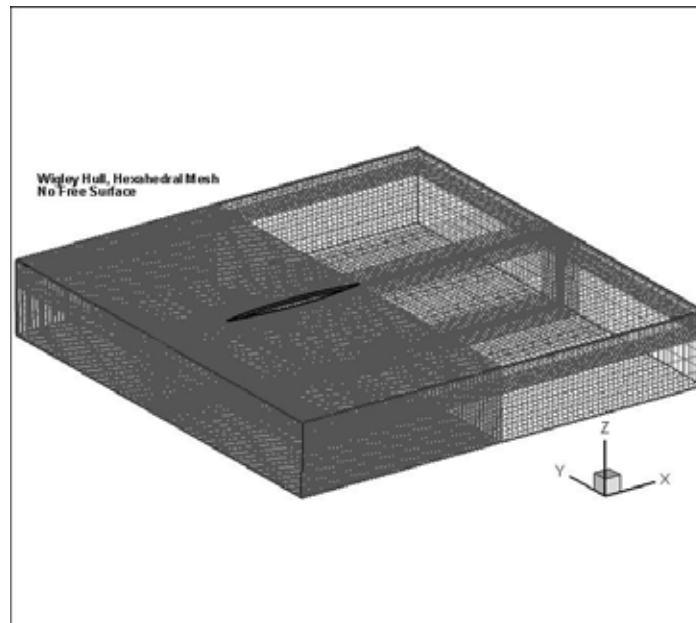
One other area that was finely meshed was the region just below the bottom of the hull, below the keel. This was done in an attempt to achieve accurate predictions of the velocity vectors in the region where the flow was expected to separate from the hull when a yaw angle was introduced. The final free surface mesh had boundaries of  $-1 \leq x \leq 2$ ,  $-1.5 \leq y \leq 1.5$ ,  $-0.5 \leq z \leq 0.05$  for a hull with an overall length of 1.0. A total of 870 578 elements made up the entire domain. For the case without a free surface, the boundaries

were  $-1 \leq x \leq 2$ ,  $-1.5 \leq y \leq 1.5$ ,  $-0.5 \leq z \leq 0$  for a hull with an overall length of 1.0. This mesh had a total of 627 626 elements.

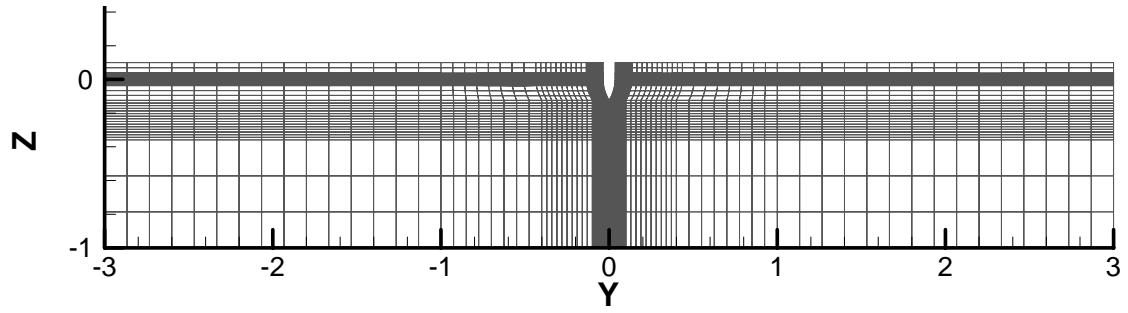
In order to compare with published data from experiments using a 2m model, the mesh was scaled within *Fluent*, using the scaling function to double the size of the domain. This scaled mesh is shown in Figures 2 and 3 below, along with representative sections parallel to the  $yz$ -plane [Figure 4].



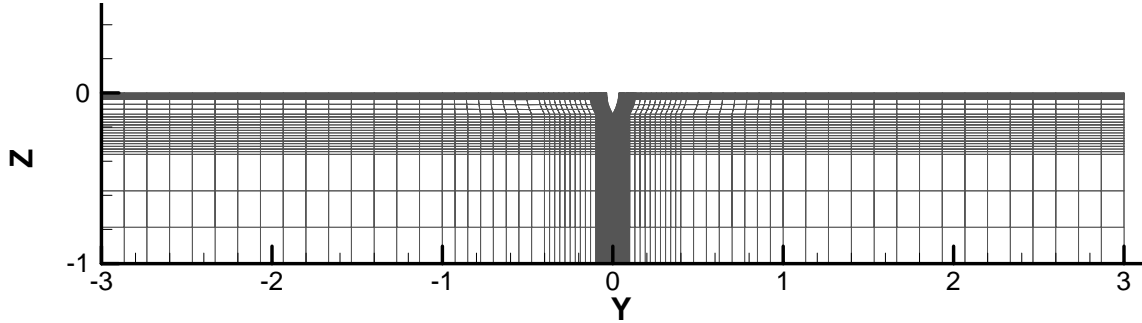
**Figure 2:** Hexahedral Mesh for Wigley Hull, origin in center of hull,  $x$ -axis positive towards stern, Free surface at  $z = 0$ .



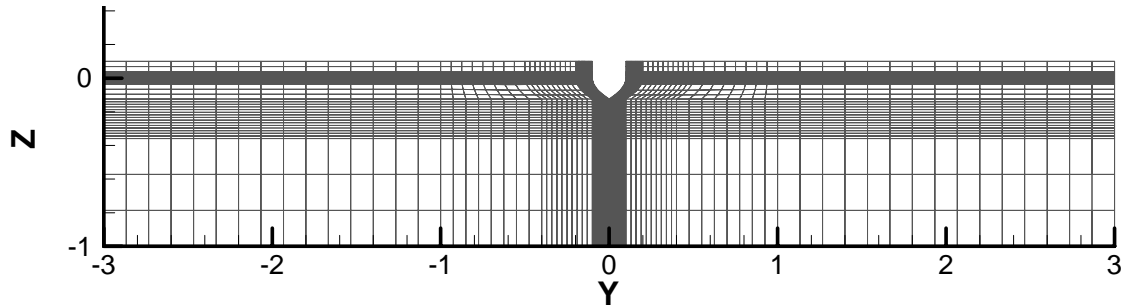
**Figure 3:** Hexahedral Mesh for Wigley Hull, origin in center of hull,  $x$ -axis positive towards stern, No free surface.



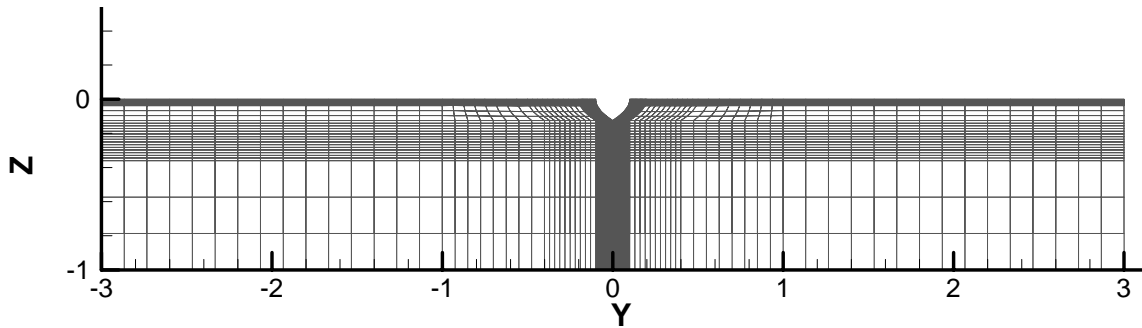
(a) Section at  $0.1L$ , free surface at  $z = 0$



(b) Section at  $0.1L$ , no free surface



(c) Section at  $0.5L$ , free surface at  $z = 0$



(d) Section at  $0.5L$ , no free surface

**Figure 4:** Representative Sections of hexahedral meshes, free surface case shown in (a) and (c), no free surface case shown in (b) and (d)



## CFD SOLUTIONS OBTAINED USING *FLUENT*

For the free surface case, the upstream end of the domain was defined as a pressure inlet, while the downstream end was defined as a pressure outlet. The hull surface was defined as a no-slip wall, and the side, lower, and upper surfaces were defined as walls with zero shear force. For cases with non-zero yaw angle, the upstream side was changed to a pressure inlet, the downstream side to a pressure outlet. For the case with no free surface, the conditions are the same, except the inlet(s) are defined as velocity inlets rather than pressure inlets.

*Fluent* was used to obtain predictions of the flow and  $x$ -,  $y$ - and  $z$ -forces at the hull. Flow entered the domain through a pressure inlet on the upstream boundaries (or velocity inlet in the case of no free surface) and exited through a pressure outlet on the downstream boundaries. Yaw angles were controlled by changing the  $x$  and  $y$  components of the direction vector, where the cosine of the yaw angle gave the  $x$ -component and sine of the angle gave the  $y$ -component.

The multiphase model utilized was the ‘Volume of Fluid’ (VOF) model. This model is designed for situations where the interface between two (or more) immiscible fluids is of specific interest. There is a single set of momentum equations for both fluids and the volume fraction of each computational cell is tracked throughout the domain [2]. The volume fraction can lie between 1 and 0, where a value of 0 indicates Fluid 0 is present and a value of 1 indicates Fluid 1 is present. Intermediate values are indicative of the interface between the two fluids (i.e. the free surface) [3]. The standard  $k$ - $\epsilon$  turbulence model was used, applying the default settings within *Fluent*, summarized in Table 1. Turbulence intensity and turbulence viscosity ratio were set at 1% and 1 respectively. The flow was solved for the steady state case. The convergence limit for all parameters was set to  $10^{-3}$  and all solutions converged within these limits.

**Table 1:** Parameters for  $k$ - $\epsilon$  Turbulence Model

$C_\mu$	0.09
$C1-\epsilon$	1.44
$C2-\epsilon$	1.92
TKE Prandtl Number	1

## SUMMARY OF CALCULATED FORCES AND MOMENTS

The forces and moments predicted by *Fluent* are summarized in Table 2 and 3 respectively. These values were used to calculate the non-dimensional force coefficients, which are presented in later sections. Using Table 2, it can be observed that as speed increases, the magnitude of  $F_x$  and  $F_y$  increase, while  $F_z$  decreases. As the yaw angle increases from  $0^\circ$  to  $40^\circ$ ,  $F_x$  decreases, and  $F_y$  increases. The forces in the  $x$  and  $y$  direction are similar for the cases with and without the free surface. However, as the angle increases the discrepancy becomes larger. For example, at  $0^\circ$  yaw, the force in the  $x$ -direction differs by 3% but at  $30^\circ$  yaw, the difference is almost 45%.

The value of all the moments ( $M_x$ ,  $M_y$  and  $M_z$ ) increase as the speed is increased and as the yaw angle is increased.

The number of iterations required for each solution is included in Tables 2 and 3. They demonstrate the significant time commitment needed when the free surface is included. In most cases, the number of iterations for the free surface case is 4-5 times the number required without the free surface.

**Table 2: Summary of Forces**

Yaw Angle (°)	Speed (m/s)	$F_n$	# Iterations		$F_x$ (N)		$F_y$ (N)		$F_z$ (N)	
			FS	nFS	FS	nFS	FS	nFS	FS	nFS
0	1.18266	0.267	1031	205	2.2427	2.1772	-0.0094	0.0002	207.650	-7.4025
0	1.32883	0.300	1050	200	2.6880	2.7064	-0.0174	0.0011	204.020	-9.3462
0	1.39971	0.316	983	197	3.1254	2.9842	-0.0191	0.0010	201.426	-10.3733
10	1.18266	0.267	1045	200	2.2670	2.1677	18.977	18.811	199.938	-13.630
10	1.39971	0.316	1047	291	3.4709	2.8293	27.312	26.890	193.922	-18.559
15	1.18266	0.267	1067	207	2.0334	2.0720	35.292	35.447	195.979	-19.833
20	1.18266	0.267	1314	250	2.0135	1.8823	53.874	57.515	192.310	-27.024
30	1.18266	0.267	1645	609	1.9054	0.8516	89.267	112.83	181.080	-35.329
40	1.18266	0.267	1675	935	1.7097	-1.101	119.52	155.43	173.615	-38.070

**Table 3: Summary of Moments**

Yaw Angle (°)	Speed (m/s)	$F_n$	# Iterations		$M_x$ (Roll)		$M_y$ (Pitch)		$M_z$ (Yaw)	
			FS	nFS	FS	nFS	FS	nFS	FS	nFS
0	1.18266	0.267	1031	205	0.00002	-0.000002	-0.2455	-0.0299	0.002	0.0011
0	1.32883	0.300	1050	200	-0.00046	0.00002	-0.9707	-0.0322	-0.006	0.0013
0	1.39971	0.316	983	197	-0.0006	-0.00002	-0.2455	-0.0332	0.002	0.0013
10	1.18266	0.267	1045	200	0.7917	1.0276	-1.339	0.216	-7.804	-5.9755
10	1.39971	0.316	1047	291	1.1237	1.4132	-1.729	0.193	-11.183	-8.627
15	1.18266	0.267	1067	207	1.4848	1.8633	-2.447	0.255	-12.261	-9.1476
20	1.18266	0.267	1314	250	2.2963	2.8756	-3.130	0.215	-17.949	-12.352
30	1.18266	0.267	1645	609	3.7904	4.7957	-5.029	-1.99	-24.775	-23.240
40	1.18266	0.267	1675	935	4.9113	6.420	-7.11	-7.89	-29.566	-38.176

FS = Free Surface

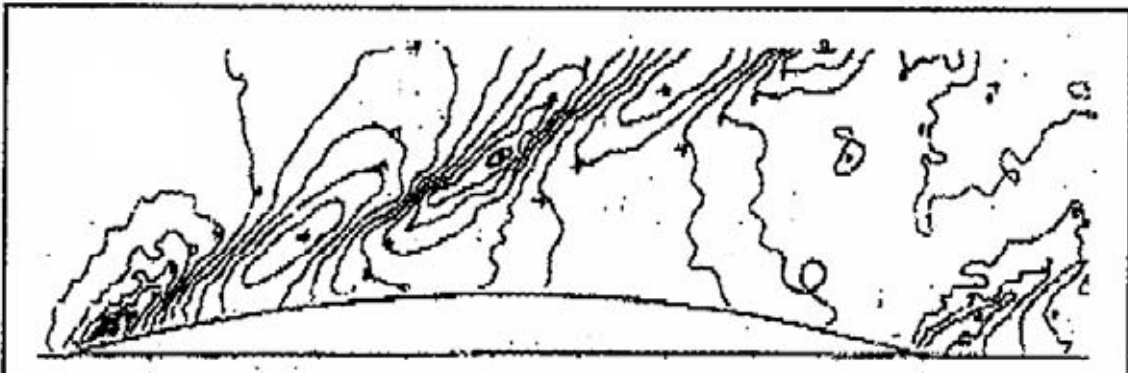
nFS = No Free Surface

## EVALUATION OF RESULTS AGAINST PUBLISHED EXPERIMENTAL AND NUMERICAL RESULTS

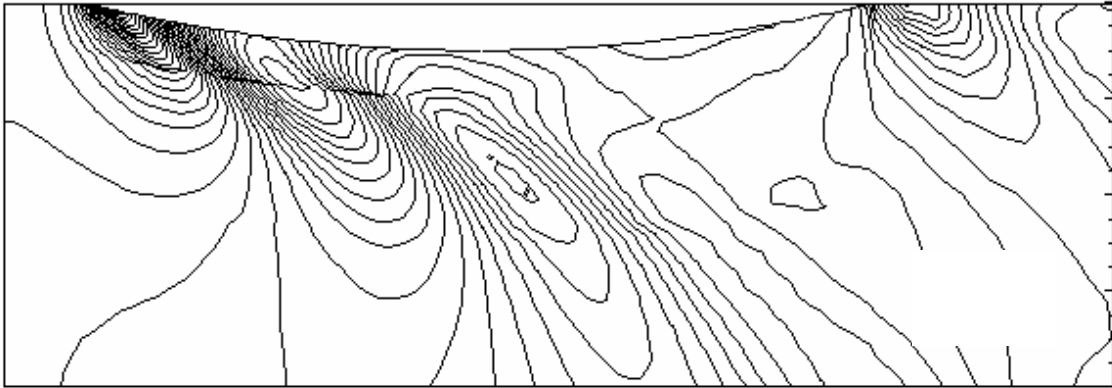
### Wave Contours

#### ZERO YAW

The wave contours at the free surface are compared with published experimental and numerical results. Figure 5 below compares the free surface contours calculated using *Fluent* with experimental data (model test, YNU [4]), for  $F_n=0.267$ . Figure 6 compares the current results with the numerical results published by Lee and Soni [5], who used a higher-order finite analytic scheme (again  $F_n=0.267$ ).

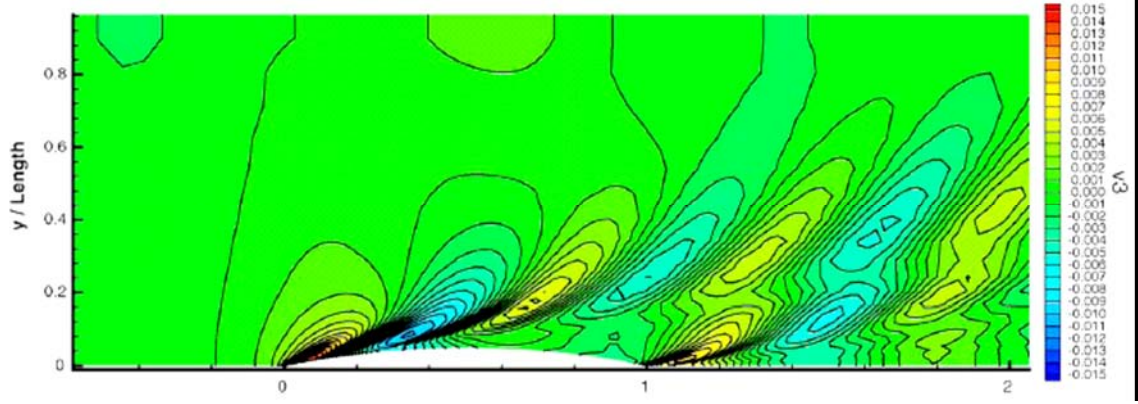


(a) Experimental wave contours (YNU Model Test)

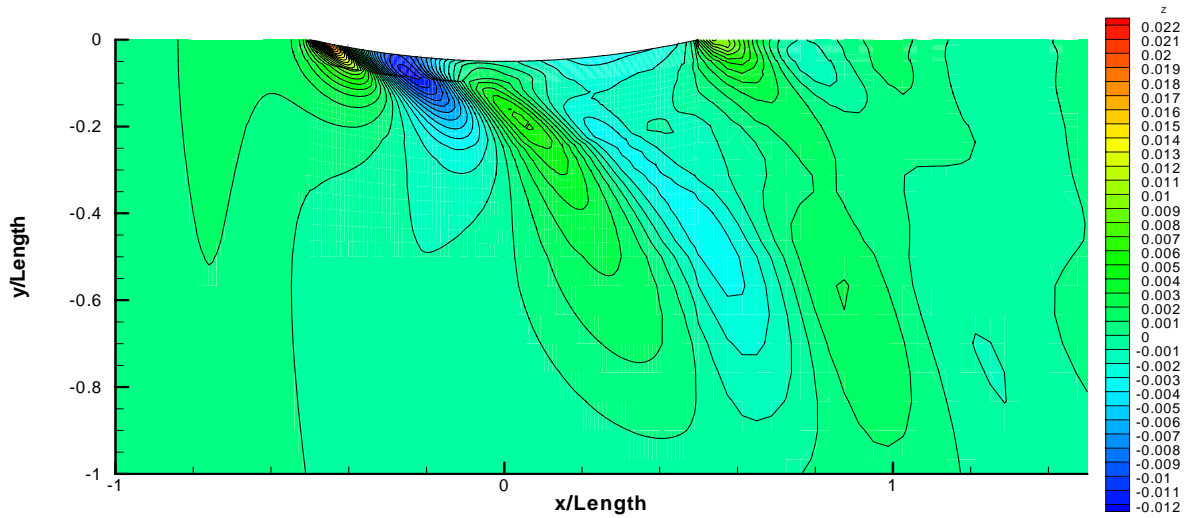


(b) Wave contours computed using *Fluent*

**Figure 5:** Comparison of *Fluent* wave contours at the free surface with YNU experimental result for  $F_n=0.267$ .



(a) Wave contours published by Lee and Soni

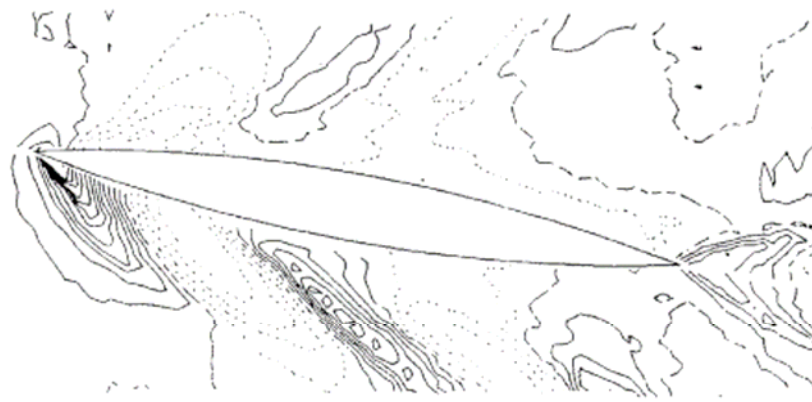


(b) Wave contours computed using *Fluent*

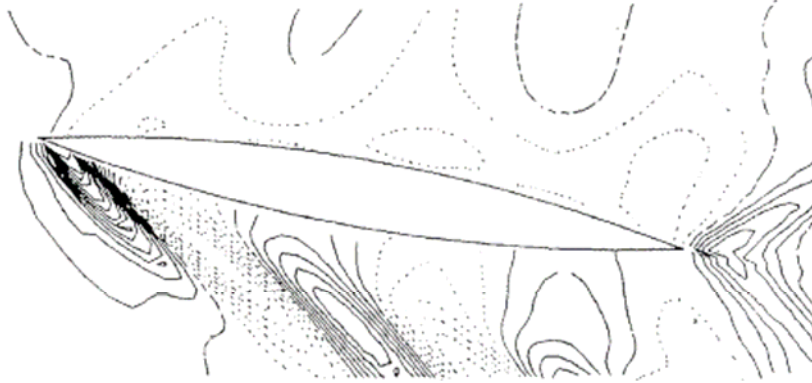
**Figure 6:** Comparison of free surface wave contours with published computational data ( $F_n = 0.267$ ,  $\alpha = 0^\circ$ )

### YAW ANGLE 10 DEGREES

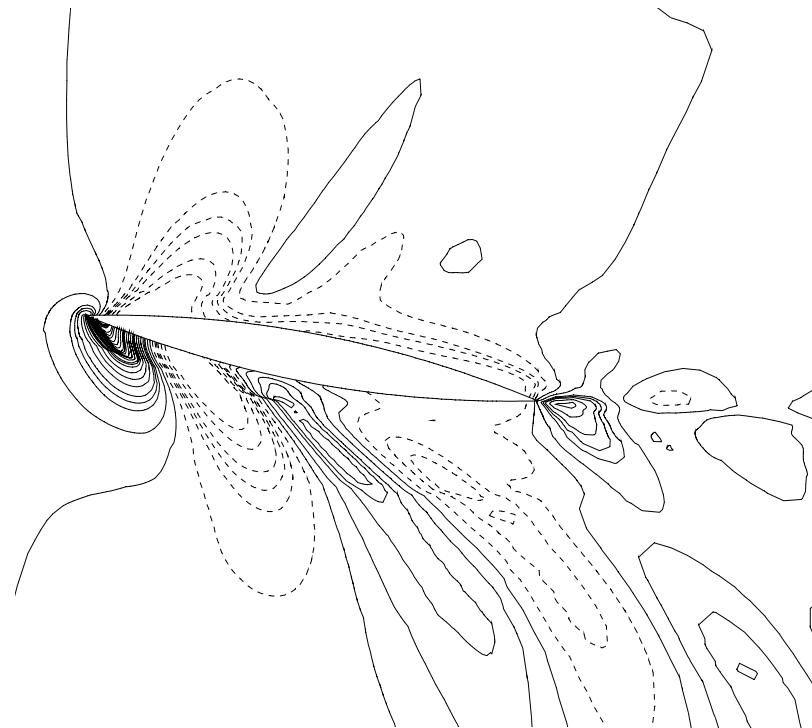
For the  $10^\circ$  yaw angle case, the wave contours are compared with experimental and computational data [6] in Figure 7. Again,  $F_n = 0.267$ , and positive and negative contours are represented by solid and dotted lines respectively. In all cases the wave contours compare quite well with published data. One possible improvement may be to refine the mesh further from the hull, as the area around the hull shows excellent agreement, but the area further away ( $>0.5L$ ) could benefit from better definition.



(a) Experiment



(b) Published non-linear Computation



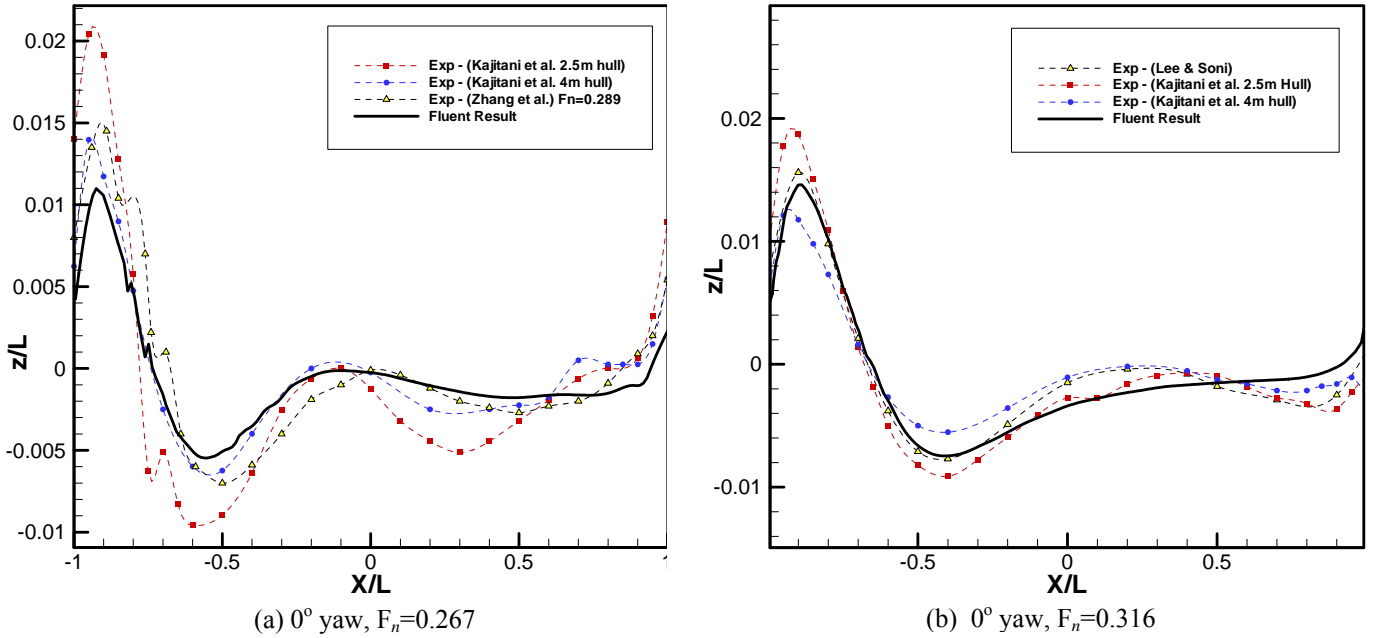
(c) Fluent Results

**Figure 7:** Wave Contours ( $Fr=0.267$ ,  $\alpha=10^\circ$ , distance between contours lines = 0.002)  
 (a) Experimental Data [6] (b) Published computational result [6] (c) Result using Fluent

## WAVE PROFILES

### ZERO YAW

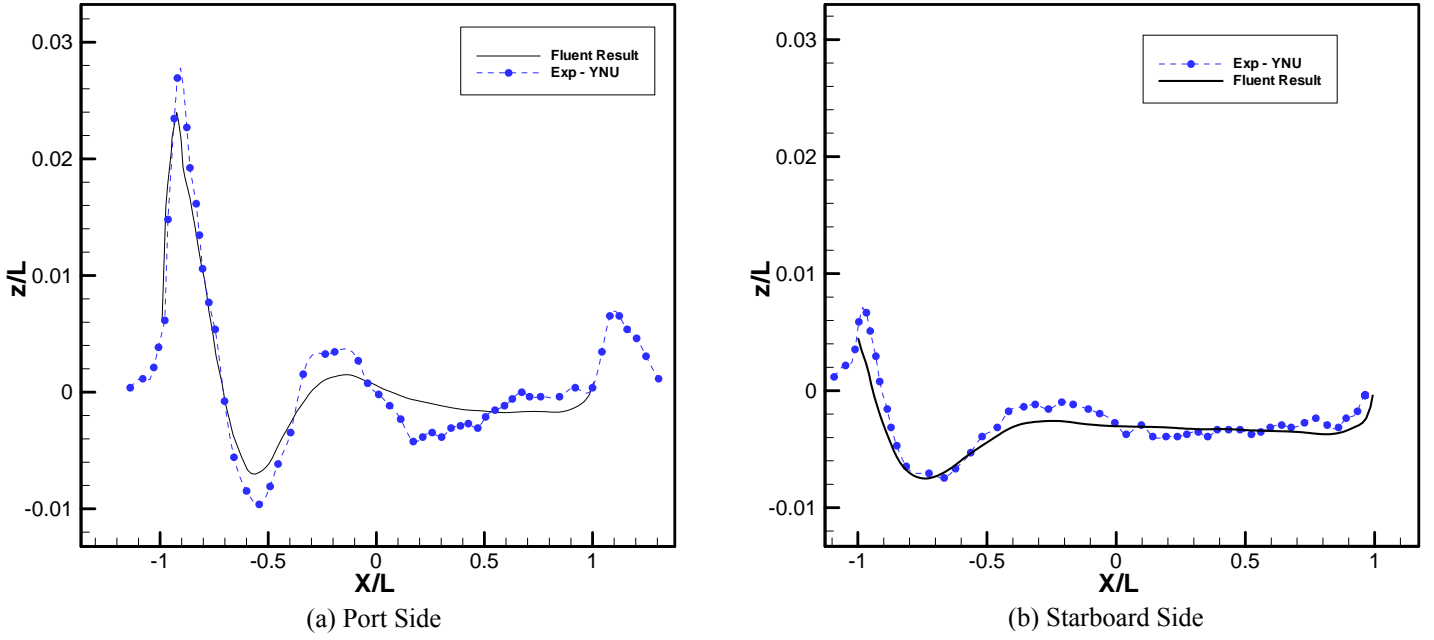
The figures below show the free-surface profiles along the hull. The profiles for 0 and 10 degree yaw angles are compared with available experimental and computational results. Figure 8(a) illustrates the profile for 0° yaw,  $F_n=0.267$ . The results obtained from *Fluent* are compared with the model data given by Kajitani et al. [7] and by Zhang [2] (note that Zhang's results are for  $F_n=0.289$ ). Figure 8(b) shows another profile for 0° yaw, but for  $F_n=0.316$ . This result is again compared to available experimental results [5],[7]. The position of maximum and minimums agree very well in both cases. For  $F_n=0.267$  the amplitude of the waves at the hull appears to be underestimated. The *Fluent* result for the higher Froude number ( $F_n=0.316$ ) shows better agreement, falling within the range of the experimental results.



**Figure 8:** Wave profiles along the Wigley hull

### YAW ANGLE 10 DEGREES

The wave profiles for 10° yaw angle ( $F_n=0.267$ ) are compared to experimental results [Figure 9] from YNU [6]. The port and starboard sides of the Wigley hull are the pressure and suction sides, respectively. Both sides show quite good agreement with the experimental results, although again, a slight underestimation of the height of the waves occurred in the *Fluent* results.



**Figure 9:** Wave profile along the Wigley hull 10° yaw,  $F_n=0.267$

### Force Coefficients

The values of coefficients can also be compared to evaluate the accuracy of results. At zero yaw angle, the most commonly used coefficient is  $C_T$ , which is calculated according to the equation

$$C_T = \frac{\text{Drag}}{\frac{1}{2} \rho V_s^2 S}$$

where  $S$  = wetted surface area. In the cases where the yaw angle is greater than zero, two coefficients can be studied – one using the x-component of the force and the other the y-component. These are referred to as longitudinal and lateral forces, respectively.

$$\text{Longitudinal Force} = \frac{F_x}{\frac{1}{2} \rho V_s^2 S} \quad \text{Lateral Force} = \frac{F_y}{\frac{1}{2} \rho V_s^2 S}$$

In general, the forces calculated using *Fluent* are higher than those reported in the literature (see Tables 4 and 5), both experimental and computational values. This is possibly caused by the absence of sinkage and trim in the *Fluent* model, as it is fixed within the grid domain.



**Table 4: Comparison of  $C_T$  Values ( $0^\circ$  Yaw)**

Angle	$F_n$	$C_T$		Source
		<i>Fluent</i>	Literature	
0	0.267	0.005393	0.00471	[7]
			0.00451	[7]
0	0.300	0.005120	0.00521	[8] (computational)
			0.00532	[8] (Model Test)
			0.00517	[9] (computational)

**Table 5: Comparison of Coefficients ( $10^\circ$  Yaw)**

Angle	$F_n$	Coefficient	<i>Fluent</i>	Literature	Source
10	0.267	Lateral (linear)	0.04563	0.03319	[6] (Computational)
		Lateral (nonlinear)		0.03651	[6] (Computational)
		Longitudinal (linear)	0.00545	0.001254	[6] (Computational)
		Longitudinal (nonlinear)		0.000916	[6] (Computational)

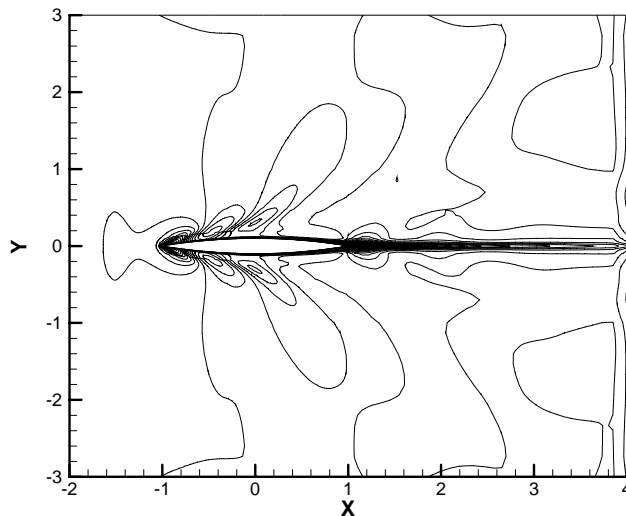
Table 4 suggests that the force coefficients are more accurate at higher Froude numbers. At  $F_n=0.267$ , the predicted value differs from the literature values by 15% and 20%. In contrast, at  $F_n=0.300$ , the predicted value is within 4% of the model test data. At  $10^\circ$  yaw, the values are quite different than the published numerical values, differing by at least 25% for the lateral coefficient. For the longitudinal coefficient, the value predicted by *Fluent* is over four times greater than the computational value published by Tahara and Longo [6]. Unfortunately, no model data is available to provide a true measure of accuracy.

## FREE SURFACE EFFECTS

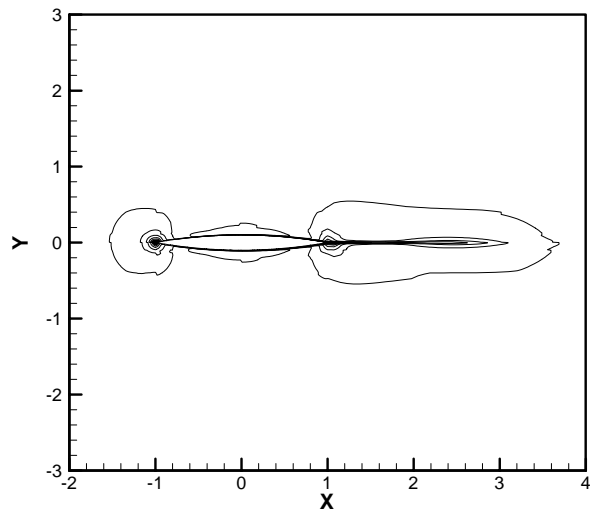
The free surface is particularly important in marine technology as ships operate at the interface between air and water. Often however, this surface has been omitted in CFD studies, as it complicates the design of the mesh, and it greatly increases the time and computer power required to obtain a solution. How much does inclusion of the free surface in the problem change the CFD predictions? Is it worth the extra time and effort, or are the predictions so similar to the case without a free surface that it is justifiable to omit it? Using the commercial program *Fluent*, the predicted velocity magnitude contours, force coefficients and velocity fields are compared for the two cases (with and without free surface). The mesh used for the no free surface case is identical to the free surface case, except that the volumes above the water line ( $z=0$ ) are removed. Identical flow conditions are used for both cases. The cases are compared at several different yaw angles to investigate the influence of yaw angle on the predictions for forces, contours and velocity fields with and without the free surface included.

### Velocity Magnitude Contours

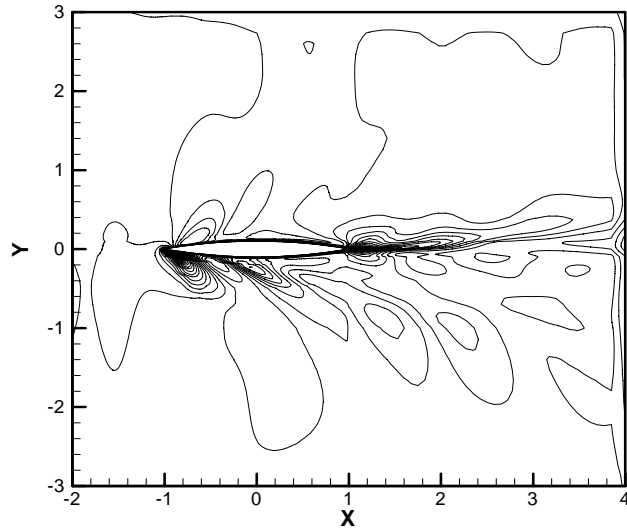
One immediately observable advantage of including the free surface in CFD studies, is the ability to plot the wave height contours at the free surface, as done previously in Figures 5 to 7. When no free surface is present, this is of course impossible. We can, however, compare other contours. Particularly of interest are the velocity magnitude contours. In the cases including a free surface, the velocity magnitude at the free surface is shown. For the cases with no free surface, the contours are shown for the top of the domain, which would be the surface of the water.



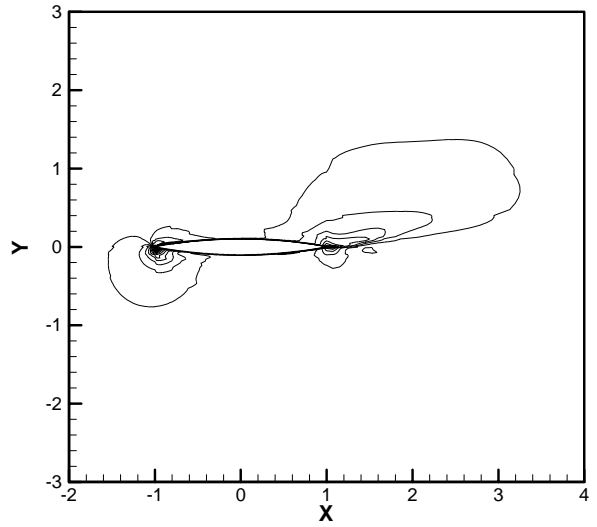
(a) Free Surface,  $0^\circ$  Yaw Angle  
Contour Range = (0.94-1.24)m/s, Interval = 0.02 m/s



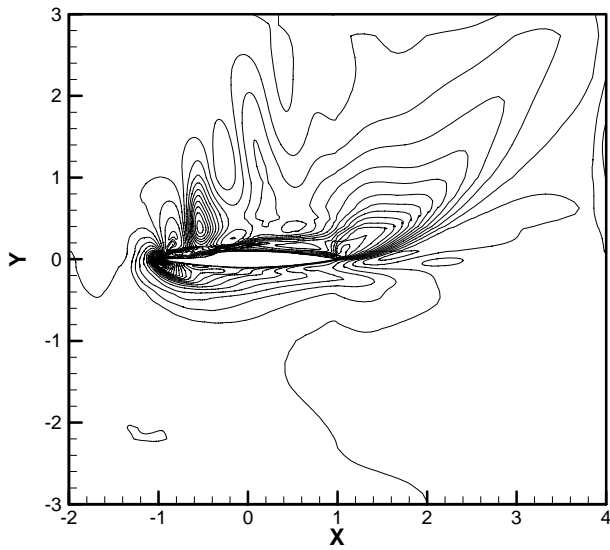
(b) No Free Surface,  $0^\circ$  Yaw Angle  
Contour Range = (0.94-1.24)m/s, Interval = 0.02 m/s



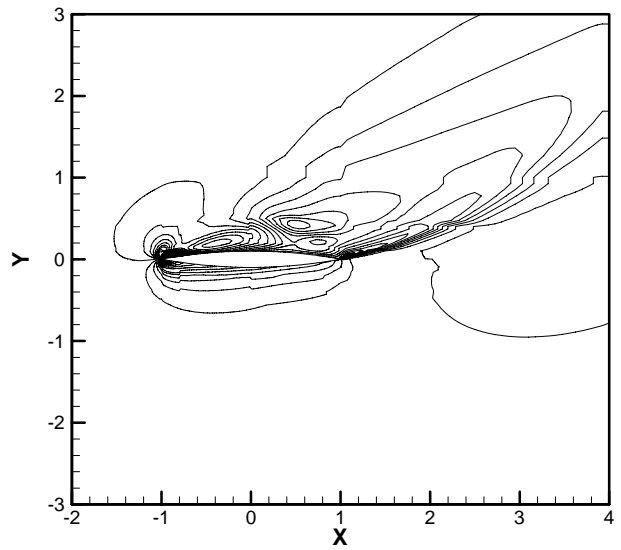
(c) Free Surface, 10° Yaw Angle  
Contour Range = (0.6-1.24)m/s, Interval =0.02 m/s



(d) No Free Surface, 10° Yaw Angle  
Contour Range = (0.6-1.24)m/s, Interval =0.02 m/s



(e) Free Surface, 40° Yaw Angle  
Contour Range = (0.1-1.6)m/s, Interval =0.05 m/s



(d) No Free Surface, 40° Yaw Angle  
Contour Range = (0.1-1.6)m/s, Interval =0.05 m/s

**Figure 10:** Velocity magnitude contours with and without a free surface for 0°, 10° and 40° yaw angle. ( $F_n=0.267$ )

The comparisons in Figure 10 above show a clear difference between the calculated velocity magnitude contours with and without the free surface. The free surface calculation shows a larger range of contours and a larger number of distinct contour levels.

## Force Coefficients

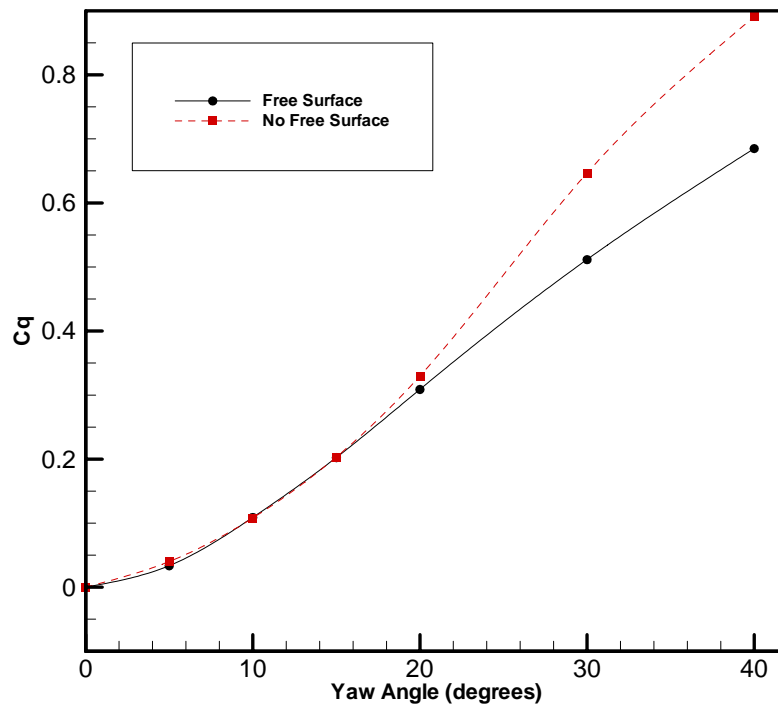
Tables 2 and 3 summarize the forces and moments predicted by *Fluent* for each set of conditions, and Tables 4 and 5 compared the calculated values of  $C_T$  to published experimental and numerical values. In this section, the influence of yaw angle on the values of coefficients will be examined. As mentioned, no model data is available for comparison. The coefficients are calculated using the expressions below, which differ slightly from those presented previously:

$$C_q = \frac{F_y}{\frac{1}{2} \rho V^2 A_L}$$

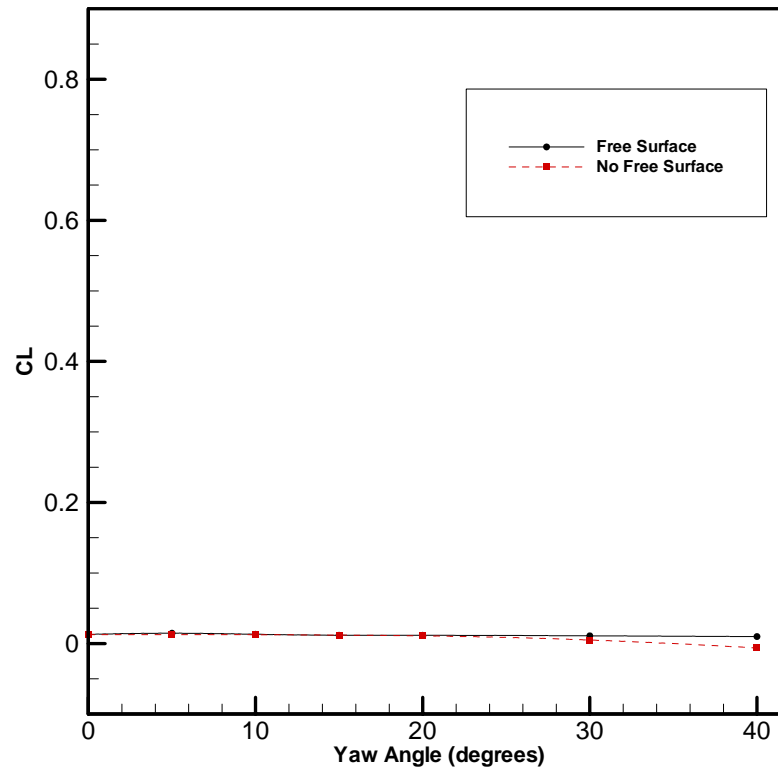
$$C_L = \frac{F_x}{\frac{1}{2} \rho V^2 A_L}$$

where  $A_L = L \times T = \text{length} \times \text{draft}$ .

Figure 11 shows the variation of these coefficients with yaw angle. The numerical values with and without the free surface are also compared in Tables 6 and 7. As expected,  $C_q$  rises as the yaw angle (and thus the force in the y-direction) increases. Beyond  $15^\circ$ , the values of  $C_q$  are higher when the free surface is not included. The values of  $C_L$  decrease as the yaw angle increases and the force in the x-direction at the hull becomes less and less predominant. There is a larger discrepancy between the  $C_q$  values of the two cases (with and without free surface), particularly at the larger yaw angles. Interestingly, the y-forces are lower when the free surface is included. The total force, defined as  $(F_x^2 + F_y^2)^{1/2}$  is plotted against yaw angle in Figure 12. Again the discrepancy between the two cases increases at higher yaw angles.



(a) Coefficient  $C_q$  (using  $F_y$ )



(b) Coefficient  $C_L$  (using  $F_x$ )

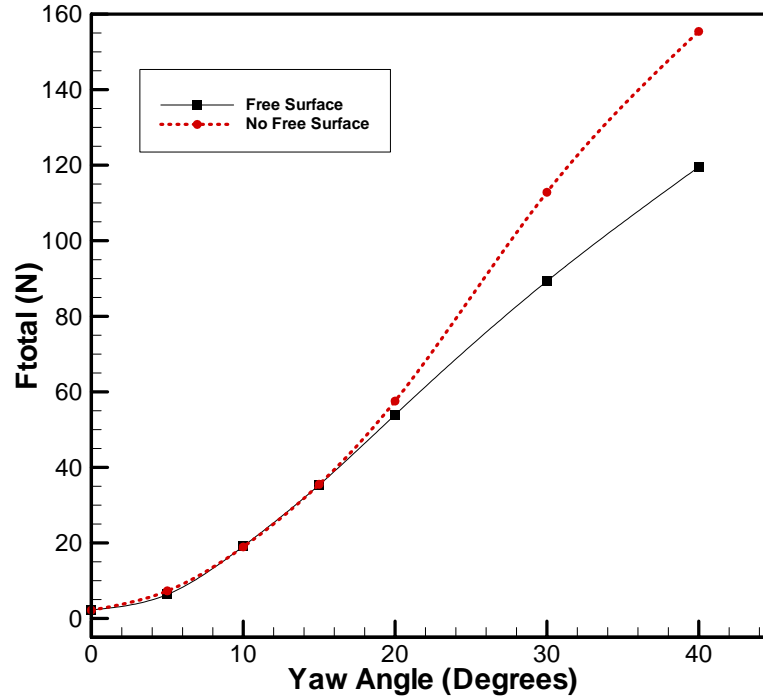
**Figure 11:** Force coefficients at increasing yaw angles ( $F_n=0.267$ )

**Table 6:  $C_L$  Values With and Without Free Surface**

Yaw Angle (°)	Speed (m/s)	$F_n$	$C_L$	
			Free Surface	No Free Surface
0	1.18266	0.267	0.01285	0.01248
0	1.32883	0.300	0.01220	0.01228
0	1.39971	0.316	0.01279	0.01221
5	1.18266	0.267	0.01457	0.01258
5	1.39971	0.316	0.01403	0.01168
10	1.18266	0.267	0.00031	0.01242
10	1.39971	0.316	0.01420	0.01157
15	1.18266	0.267	0.01165	0.01187
20	1.18266	0.267	0.01154	0.01079
30	1.18266	0.267	0.01092	0.00488
40	1.18266	0.267	0.00980	-0.00631

**Table 7:  $C_q$  Values With and Without Free Surface**

Yaw Angle (°)	Speed (m/s)	$F_n$	$C_q$	
			Free Surface	No Free Surface
0	1.18266	0.267	-5.41E-05	1.24E-06
0	1.32883	0.300	-7.90E-05	4.81E-06
0	1.39971	0.316	-7.81E-05	3.91E-06
5	1.18266	0.267	0.03334	0.03985
5	1.39971	0.316	0.03295	0.03946
10	1.18266	0.267	0.10874	0.10778
10	1.39971	0.316	0.11172	0.11000
15	1.18266	0.267	0.20222	0.20311
20	1.18266	0.267	0.30869	0.32955
30	1.18266	0.267	0.51149	0.64651
40	1.18266	0.267	0.68486	0.89061

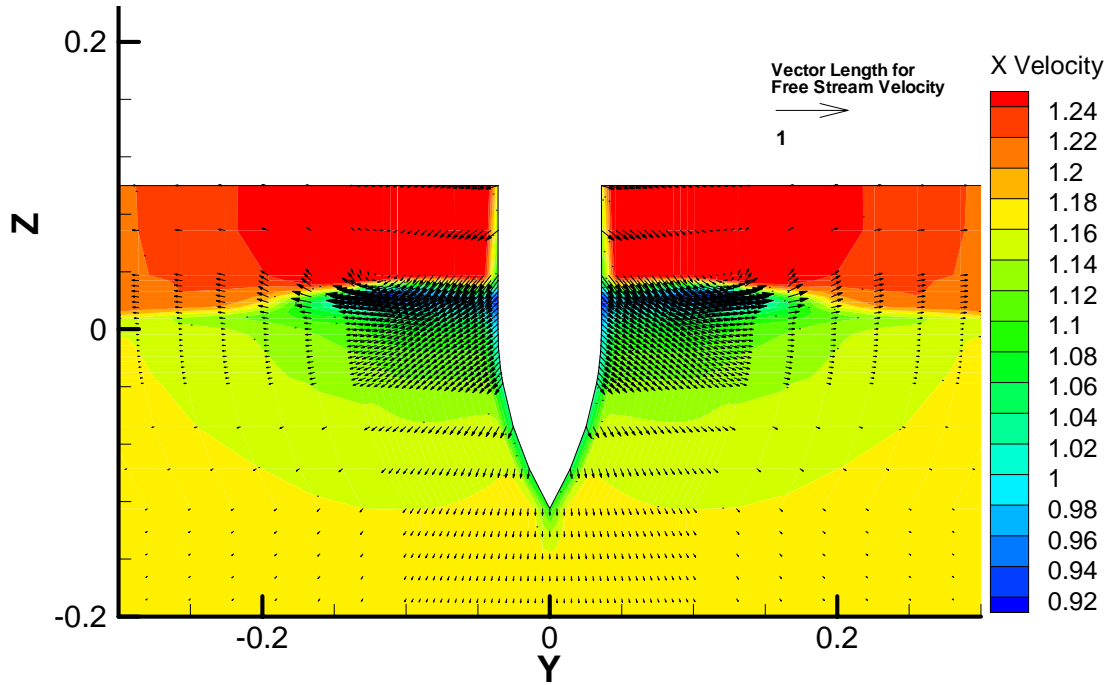


**Figure 12:** Total Force at increasing yaw angles ( $F_n=0.267$ )

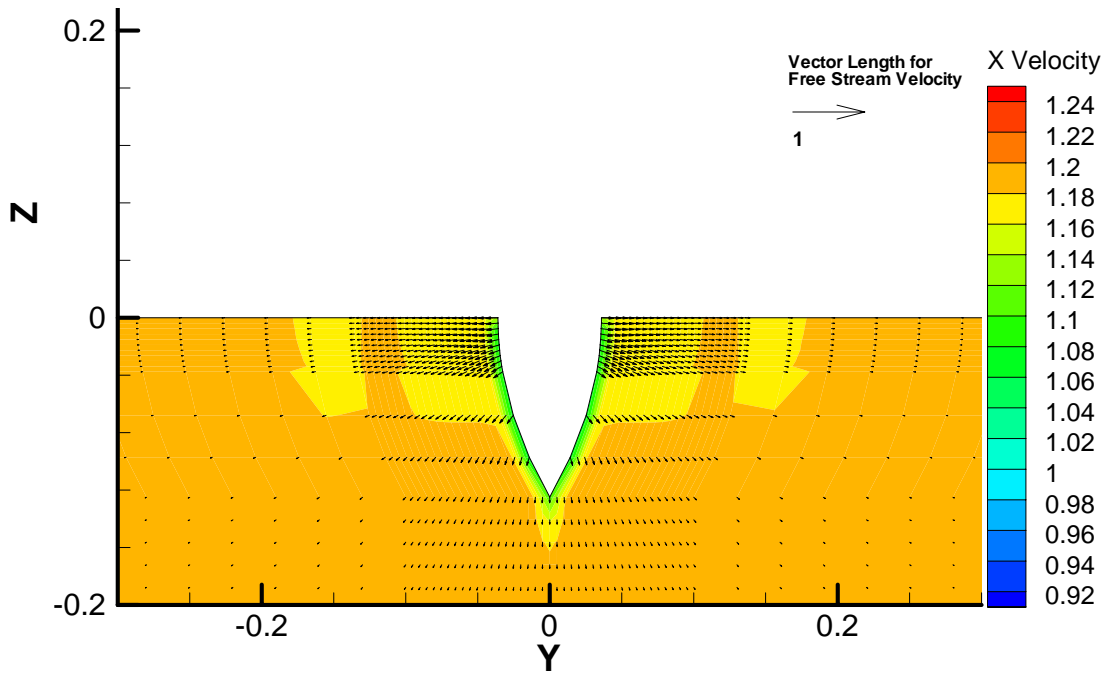
## VELOCITY FIELDS

The flow fields for several transverse planes are shown in Figures 13 to 24. The vector variables are the y- and z-velocities, while the contours represent velocity in the x-direction. As observed before, the velocity contours are much more well-defined and varied when the free surface is included. This is particularly due to the presence of two different fluids, the velocity changes quite rapidly in the air domain. Even within the water domain however, differences are observed. The x-velocity contours show the same trend as seen in the previous velocity magnitude contours and z-contours: Without the free surface, there is much less definition and a smaller range of contours throughout the field.

At  $0^\circ$  yaw,  $x/L=0.1$  (Figure 13 and 14) the velocity vectors up to and including  $z=0$  (the approximate free surface) are virtually the same for the two cases. However, for  $0^\circ$  yaw,  $x/L=0.5$ , Figures 15 and 16, the free surface predicted velocity vectors appear to be larger in magnitude than those predicted for the case without the free surface. (Note that the size of the reference vector had to be different for these two cases, in order to make the vectors visible in Figure 16, the case without a free surface). At  $10^\circ$  yaw, the vectors are still quite similar, with only minute differences appearing close to the free surface. Finally at  $40^\circ$  yaw, a few differences in the vectors are seen. These differences appear mainly on the starboard side of the hull, and again near the free surface. It appears that including the free surface does not have a noticeable effect on the velocity vectors except very near the free surface (approximately  $\leq 0.05L$ ).

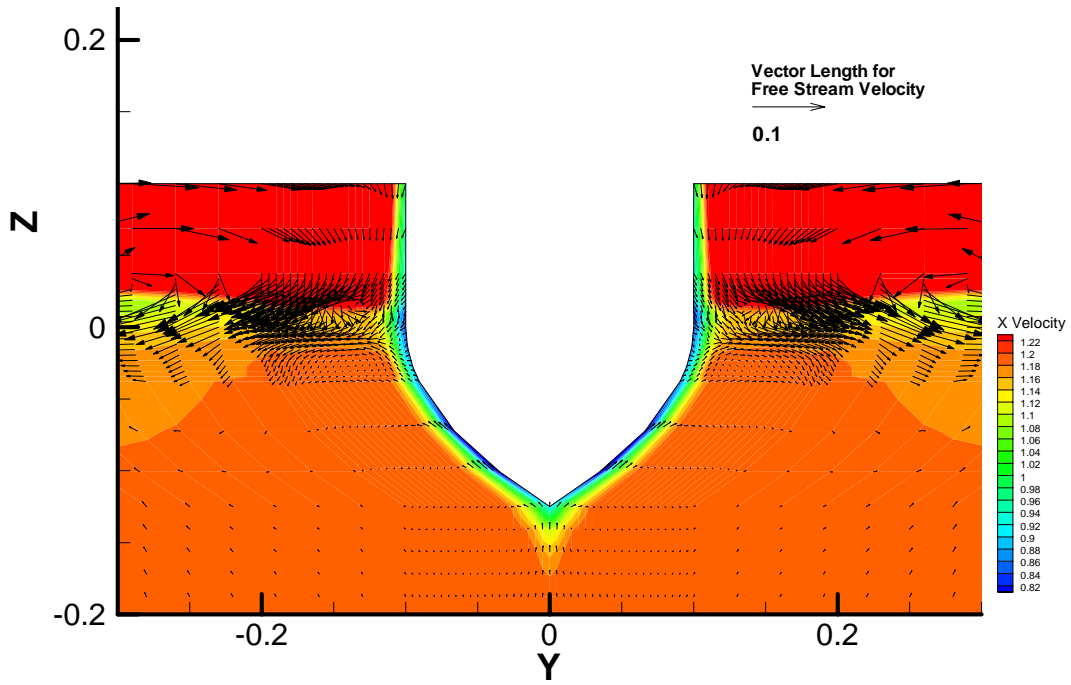


**Figure 13:** Calculated velocity field at  $x/L = 0.1$ , Free Surface ( $0^\circ$  yaw angle,  $F_n=0.267$ )

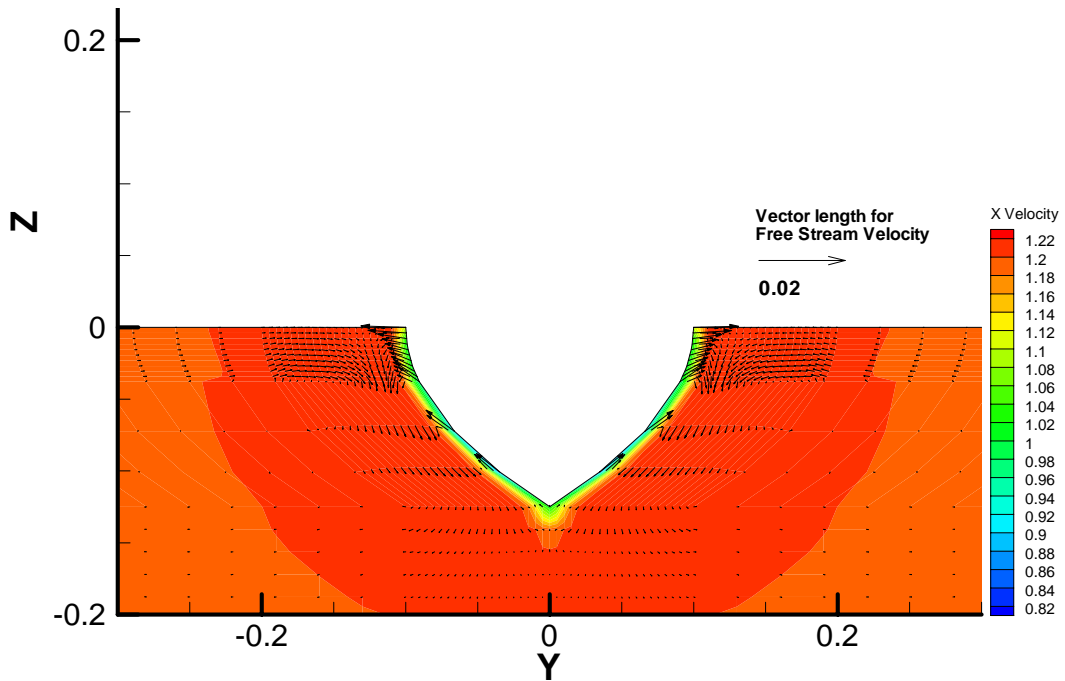


**Figure 14:** Calculated velocity field at  $x/L = 0.1$ , No free surface ( $0^\circ$  yaw angle,  $F_n=0.267$ )

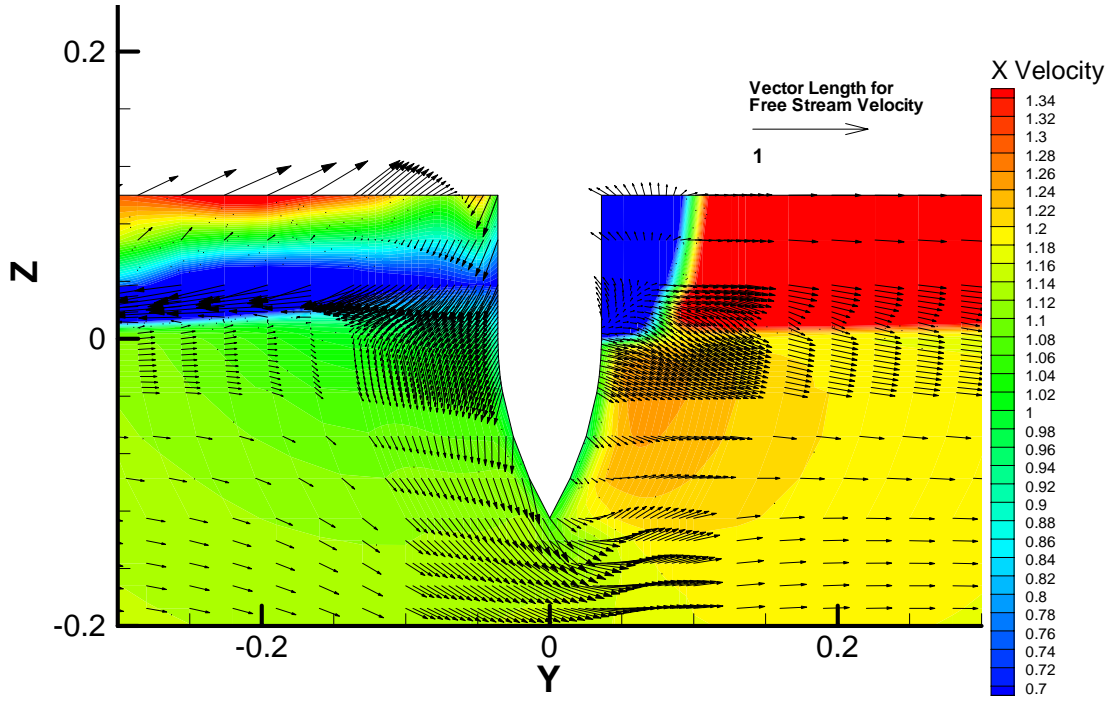




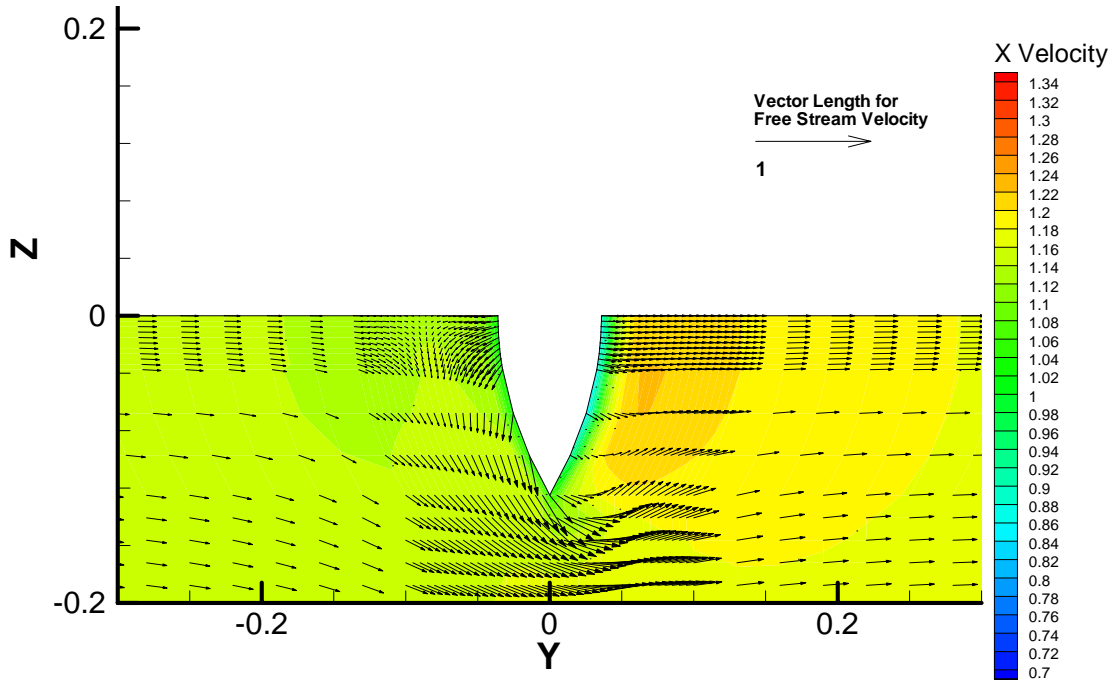
**Figure 15:** Calculated velocity field at  $x/L = 0.5$ , Free Surface ( $0^\circ$  yaw angle,  $F_n=0.267$ )



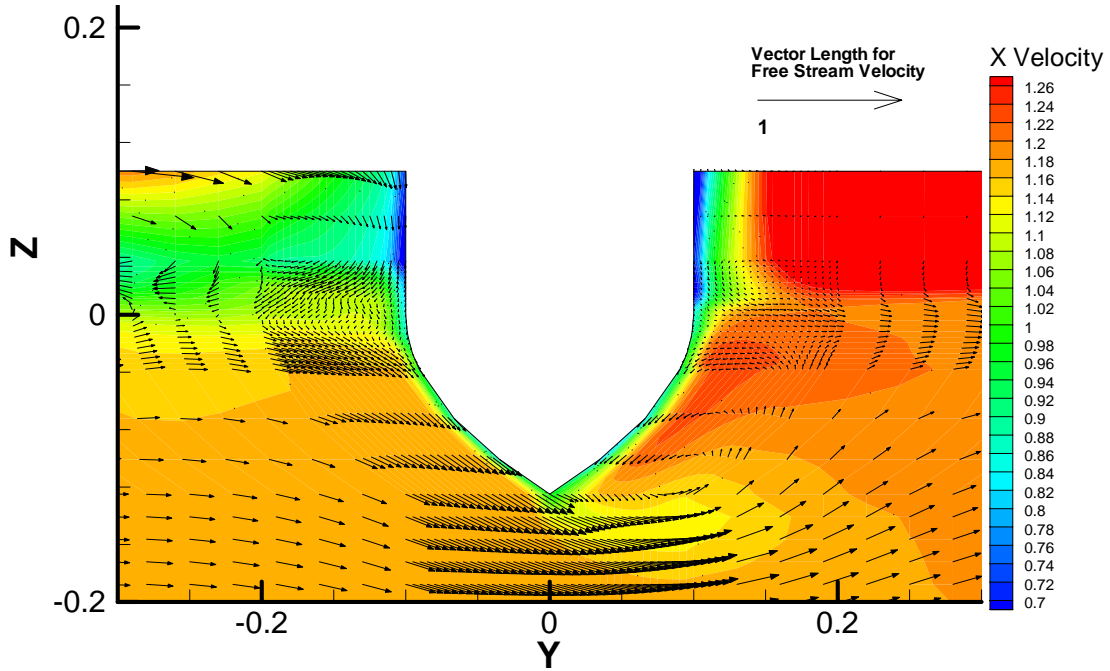
**Figure 16:** Calculated velocity field at  $x/L = 0.5$ , No free surface ( $0^\circ$  yaw angle,  $F_n=0.267$ )



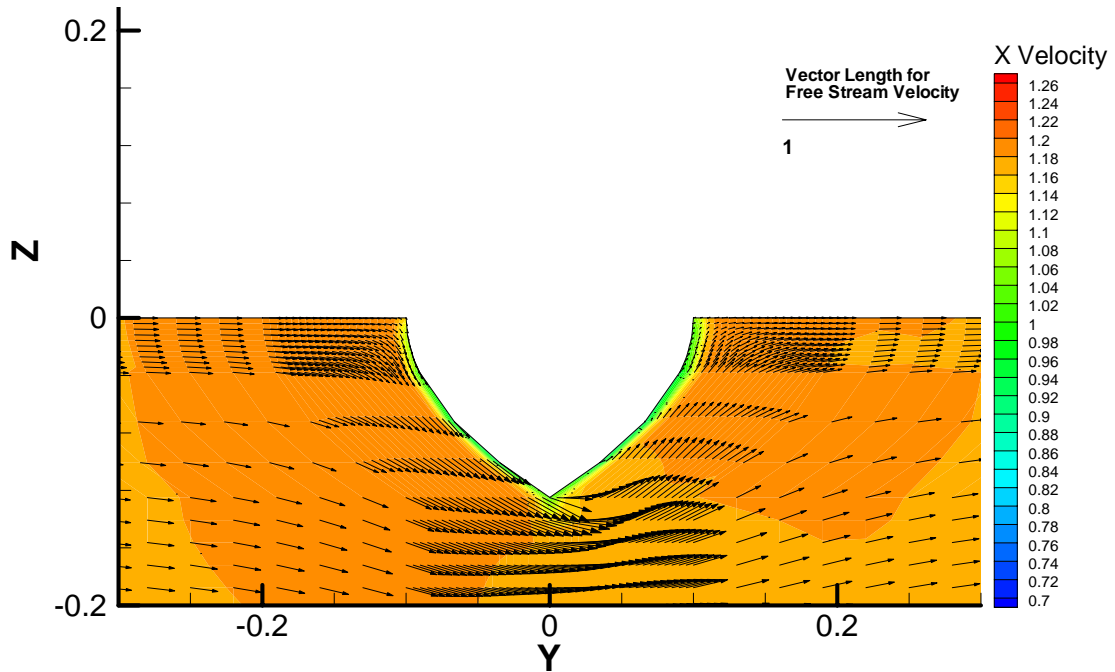
**Figure 17:** Calculated velocity field at  $x/L = 0.1$ , Free surface ( $10^\circ$  yaw angle,  $F_n=0.267$ )



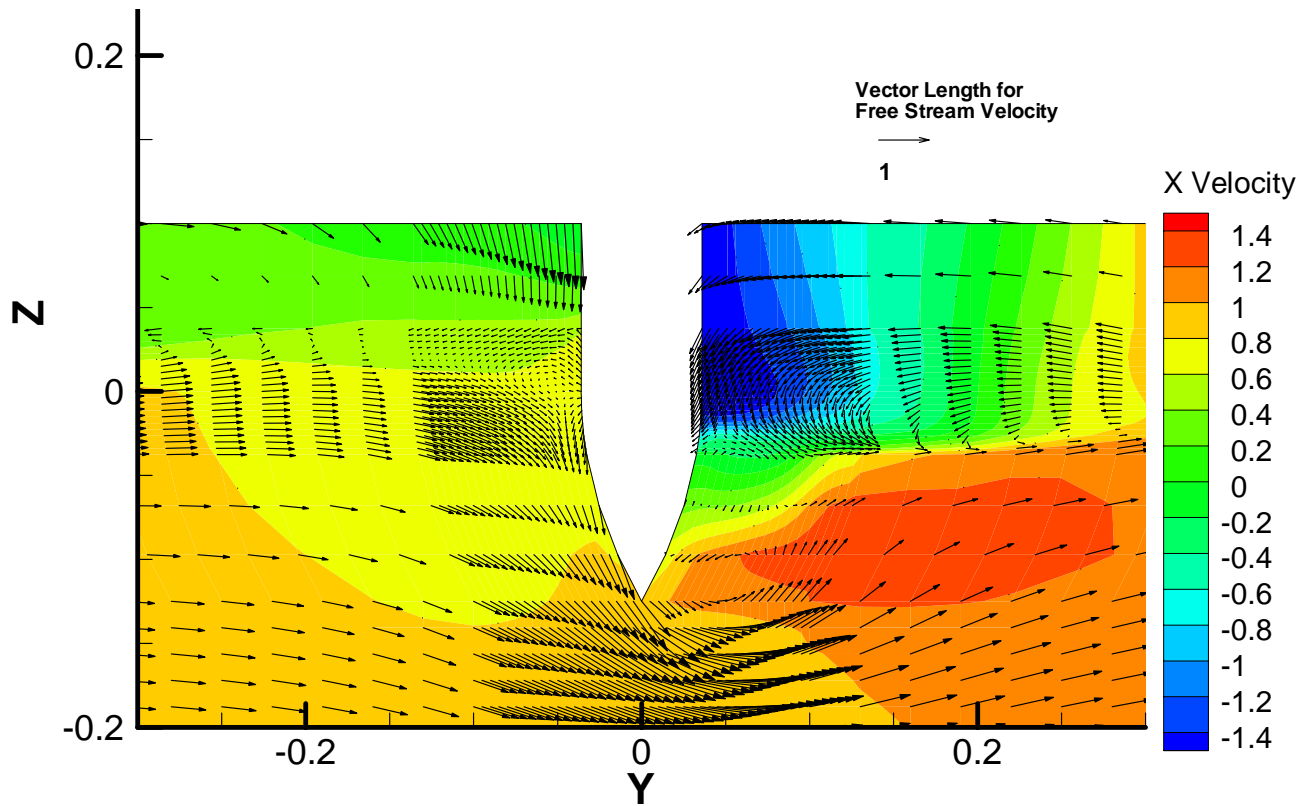
**Figure 18:** Calculated velocity field at  $x/L = 0.1$ , No free surface ( $10^\circ$  yaw angle,  $F_n=0.267$ )



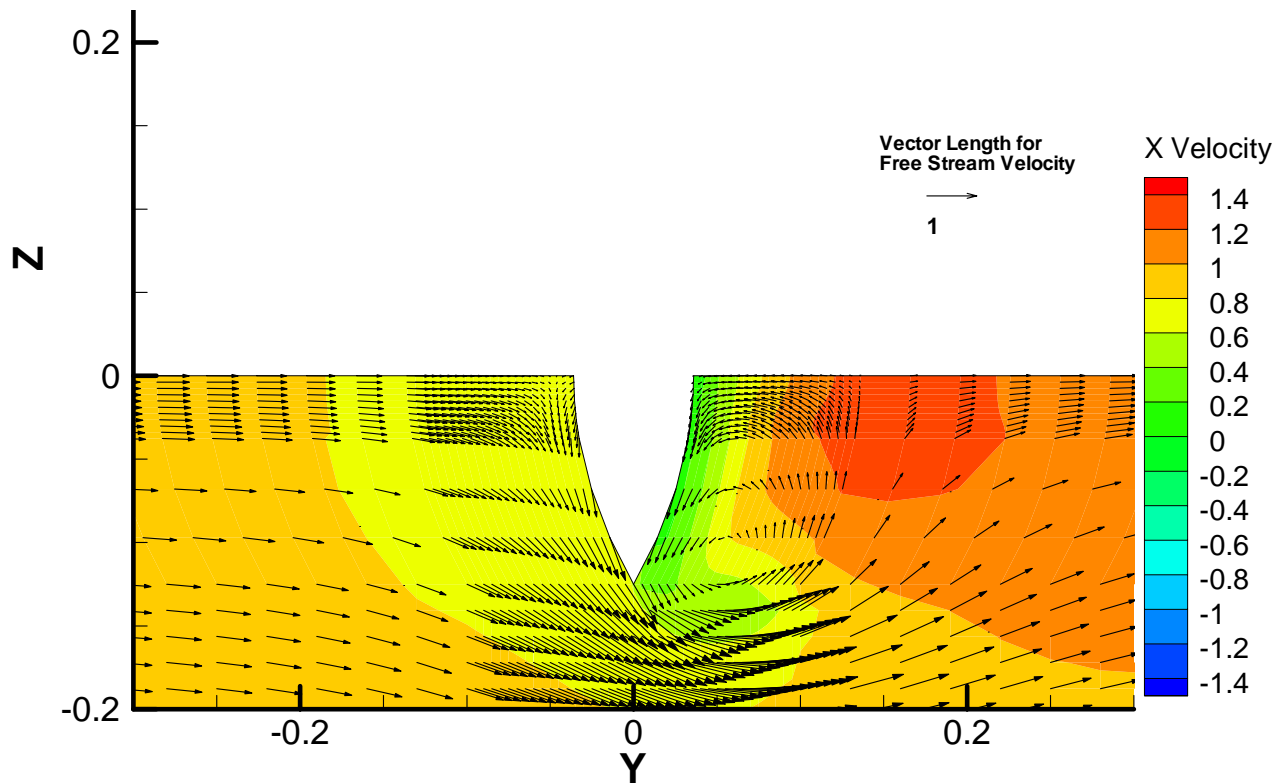
**Figure 19:** Calculated velocity field at  $x/L = 0.5$ , Free surface ( $10^\circ$  yaw angle,  $F_n=0.267$ )



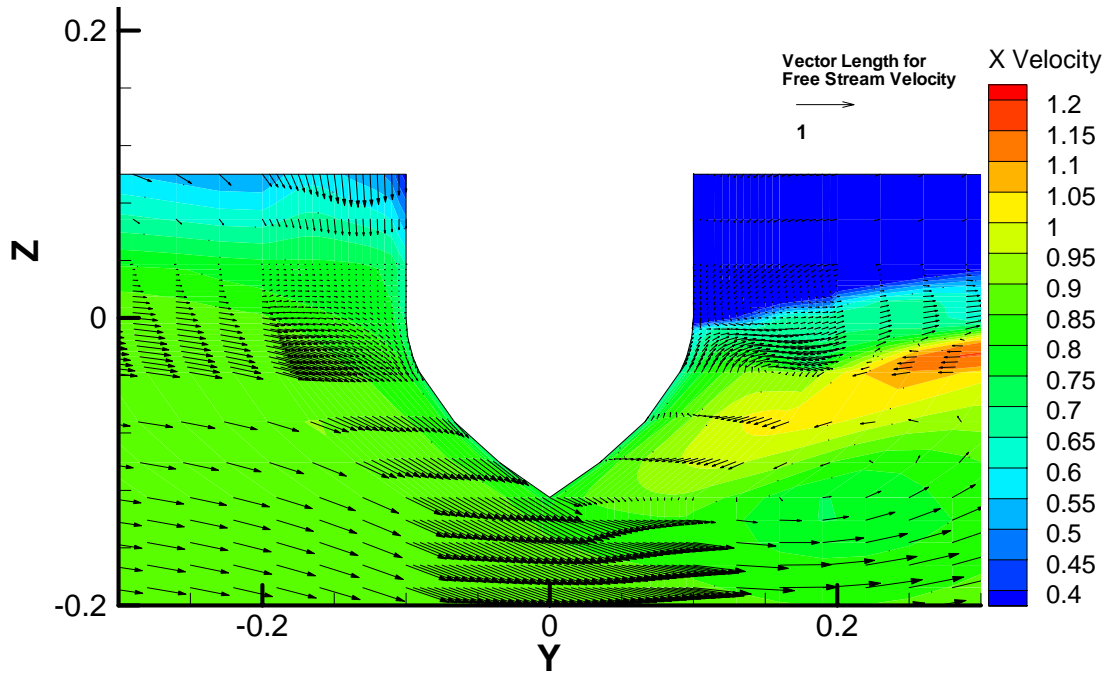
**Figure 20:** Calculated velocity field at  $x/L = 0.5$ , No free surface ( $10^\circ$  yaw angle,  $F_n=0.267$ )



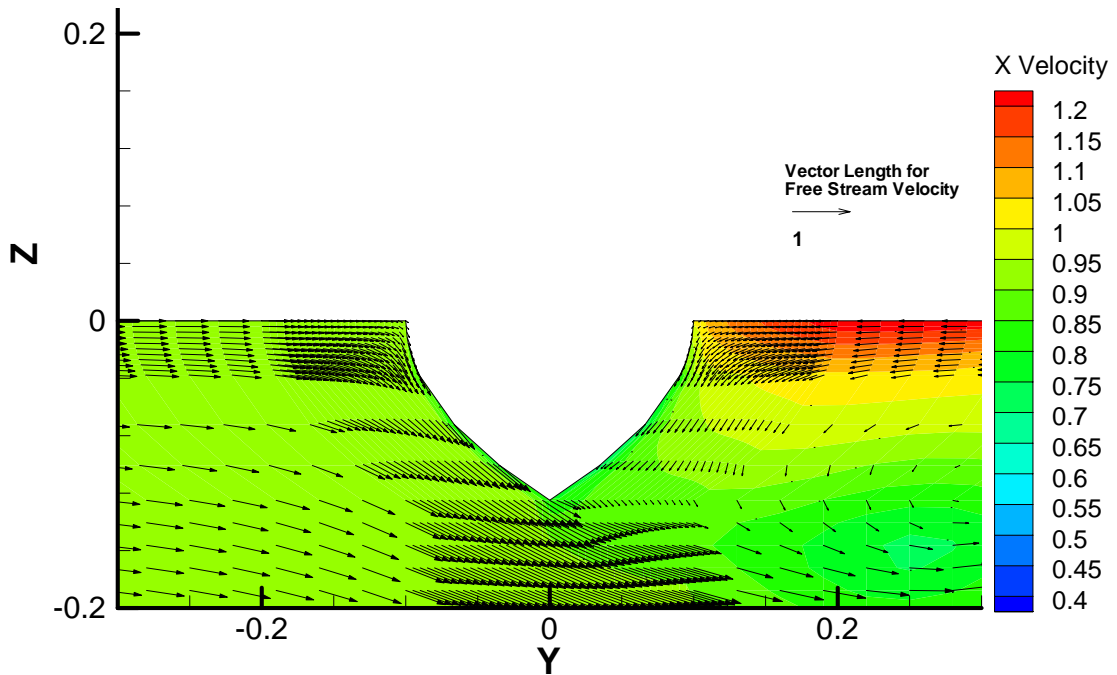
**Figure 21:** Calculated velocity field at  $x/L = 0.1$ , Free surface ( $40^\circ$  yaw angle,  $F_n=0.267$ )



**Figure 22:** Calculated velocity field at  $x/L = 0.1$ , No free surface ( $40^\circ$  yaw angle,  $F_n=0.267$ )



**Figure 23:** Calculated velocity field at  $x/L = 0.5$ , Free surface ( $40^\circ$  yaw angle,  $F_n=0.267$ )

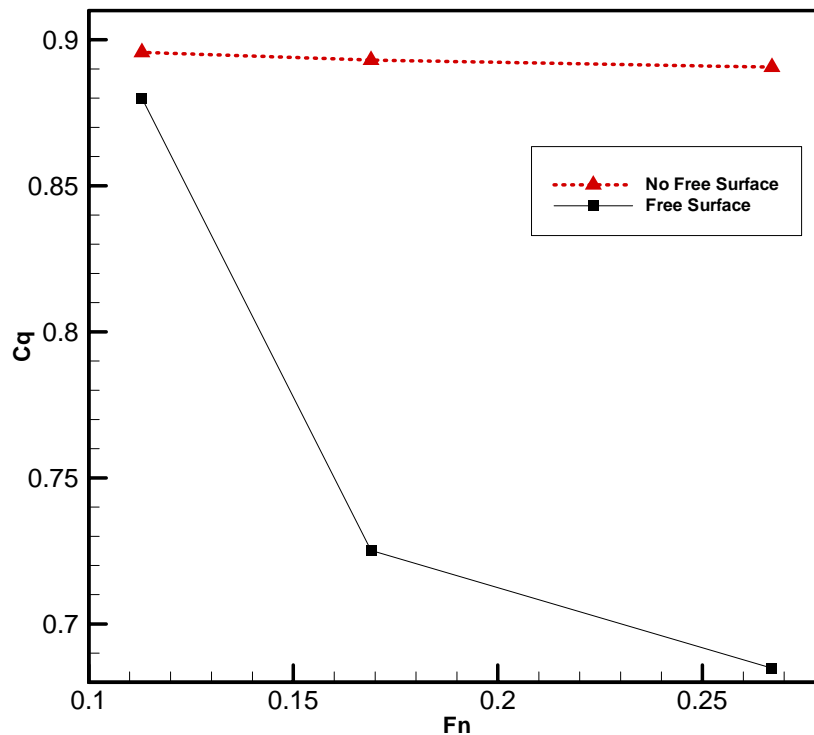


**Figure 24:** Calculated velocity field at  $x/L = 0.5$ , No free surface ( $40^\circ$  yaw angle,  $F_n=0.267$ )

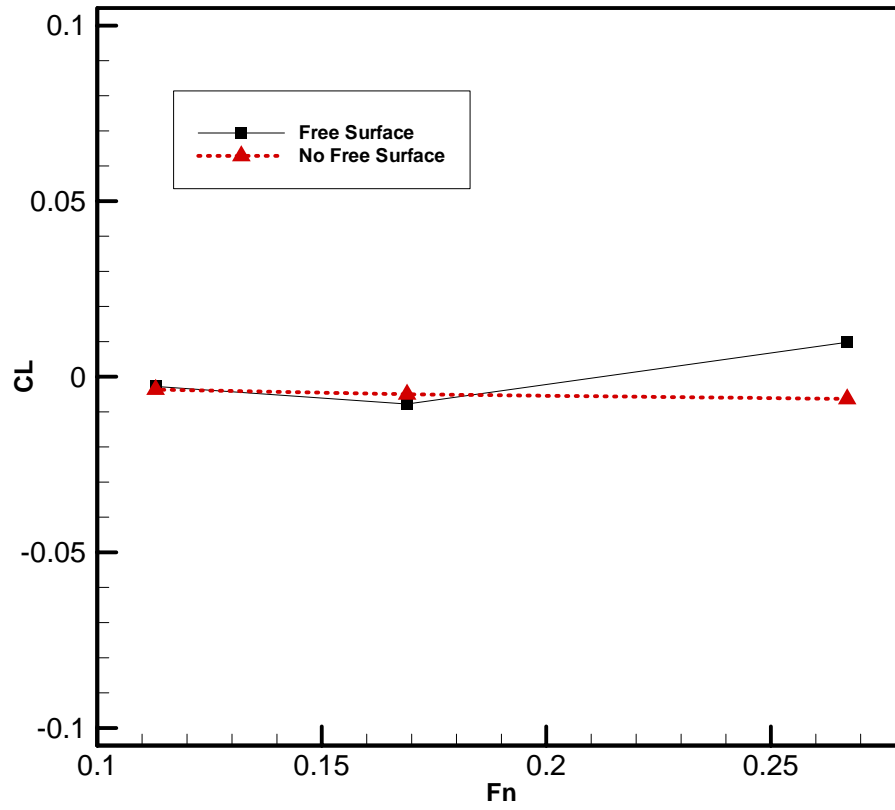
## EFFECT OF FROUDE NUMBER ON DISCREPANCY BETWEEN FREE SURFACE CASE AND NO FREE SURFACE CASE AT 40° YAW

It was observed in Figures 11 and 12 that the force coefficients, (particularly  $C_q$ ) showed a large discrepancy between the cases with and without the free surface at large yaw angles. Focusing on one large yaw angle (40°), the effect of Froude number on the force coefficients was studied. Without a free surface, the coefficients seem to remain reasonably constant, decreasing slightly as Froude number increases. When the free surface is included, there is a much larger effect on the coefficients as the Froude number changes. When the free surface is present, the predictions then include waves which form at the interface between air and water. At higher speeds, the waves created are larger, and thus have a larger effect on the force coefficients. As the speed decreases, the waves also get smaller, and have less effect on the force coefficients. Accordingly, as the Froude number decreases, and the wave effect lessens, the free surface case should begin to approach the case without the free surface. Figure 25 shows that the discrepancy between the two cases did decrease substantially at smaller Froude numbers.

Finally, it should be noted that as no experimental data is available, there is no way to determine which case is more accurate at the higher Froude numbers.



(a) Coefficient  $C_q$  (using  $F_y$ ) at 40° yaw angle



(b) Coefficient  $C_L$  (using  $F_x$ ) at  $40^\circ$  yaw angle

**Figure 25:** Effect of Froude number on force coefficients at  $40^\circ$  yaw.

## CONCLUSIONS

The effect of the free surface on various aspects of CFD predictions (including wave patterns, wave profiles, velocity vectors and forces) was examined. Using *Fluent*, predictions were obtained for cases with and without the free surface, particularly examining differences between the two cases at increasing yaw angles. The predictions were first compared to published results to determine their accuracy, and then compared to each other. The predictions compared favorably to experimental wave contours and wave profiles, although at lower Froude numbers, the amplitude of the waves at the hull was slightly underestimated. The predicted force values showed reasonable agreement with experiment, however, these were generally overestimated.

Including the free surface in the calculation appears to improve some predictions. The velocity magnitude contours were better defined when the free surface was included, as well as the x-velocity contours. The values of force constants were quite similar at small yaw angles, however as the angle increased, so did the discrepancy between the two cases. The velocity vectors were also very similar between the two cases, particularly at low yaw angles. The largest difference occurs at large angles, mainly near the waterline, especially on the starboard side of the hull.

Including the free surface increases the time required to create a mesh and greatly increases the length of time required to obtain a solution in *Fluent*. In cases with small yaw angles and low Froude numbers, it may be reasonable to omit the free surface, as the results are quite similar to the case without a free surface. However, if large yaw angles or high Froude numbers are required, better results are obtained by including the free surface.



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