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Experimental Evaluation of a Numerical Simulation Model for Predicting Room Air Motion

Abstract
Spatial distributions of air velocity, turbulent kinetic energy and temperature were measured in a full-scale office room (7.32 x 4.72 x 2.44 m) under non-isothermal conditions. Numerical simulation was also conducted with the computational fluid dynamics code 'EXACT3'. The numerical simulation results agreed with the measurements qualitatively but were quantitatively different. Possible reasons for these differences and future research needs are discussed in this paper.

Introduction
Predicting room air motion is important to the design and control of effective ventilation systems which can reliably provide satisfactory thermal comfort conditions and acceptable indoor air quality. With the advancement of computer technology and turbulence modelling, numerical simulation of room air and gas flow based on the computational fluid dynamics (CFD) technique has become a potential tool for designing ventilation systems. However, it is essential to validate numerical simulation models with reliable experimental data so that the simulation results can be used with confidence.

The validation task has been carried out most extensively by the International Energy Agency Annex 20. It is recognized that numerical simulation models must be validated for specific types of applications and ventilation conditions. The objective of this study was to measure detailed spatial distributions of air velocity, temperature and turbulent kinetic energy in a full-scale office room and compare the measured results with those obtained by a numerical simulation model.

Methods
Room Configuration and Test Conditions
Evaluating numerical simulation models for engineering design purposes requires well-defined test cases which represent realistic and typical ventilation conditions and room configurations. A full-scale office room with a realistic airflow rate, internal heat load and furniture arrangement was chosen for this purpose (fig. 1). In this paper, results are presented for a non-isothermal test condition with an empty room. The test conditions were (see Nomenclature for definitions of variables): \( U_0 = 3.05 \text{ m/s}, T_a = 12.8^\circ \text{C}, T_r = 23.3^\circ \text{C}, T_f = 40.0^\circ \text{C}, \Delta T_{in} = 16.7^\circ \text{C}, \text{Re}_0 = 2.016, \text{Ar}_{in} = 3.64 \times 10^{-4} \) and \( Q = 0.21 \text{ m}^3/\text{s} \).
Experimental Facility and Procedure

Experiments were conducted with a room ventilation simulator developed for room air and gas distribution studies [7, 8]. The dimensions of the full-scale office (fig. 1) were 7.32 × 4.72 × 2.44 m. The walls, ceiling and floor of the room were well-insulated with R values of 2.11, 2.11 and 3.35 °C·m²/W, respectively. Air for the test room was supplied through a continuous diffuser slot with a discharge angle of 25° from the ceiling. The diffuser air temperature was kept constant by an independent cooling unit. The internal heat loads were simulated by 41 heating panels (0.61 × 1.22 m) uniformly distributed over the floor surface. During the experiments, temperatures at the diffuser, exhaust, centre of the room and the floor surface were monitored by thermocouples and recorded with a data logger to ensure stable experimental conditions. The air temperature around the test room was maintained within ±1.5 °C of the temperature at the centre of the test room by a separate air-conditioning system, so that the heat transfer through the walls and ceiling were minimized (i.e., the adiabatic conditions at the surfaces of the walls and ceiling were well-approximated).

Room airflow patterns were visualized with neutrally thermal buoyant helium bubbles. Air velocities and temperatures inside the test room were measured with a hot-wire anemometer and resistance temperature detectors (RTD), respectively. A microcomputer-based automatic data acquisition and probe-positioning system [8] was used to collect data. The flow within the room was practically two-dimensional [9]. Measurements were taken at 22 × 41 grid points at the symmetric plane of the room to provide detailed data (fig. 2a). The uncertainties (e) involved with the experiments were as follows [8]:

- $\epsilon_{u4} = \pm 1.6\%$ or 0.015 m/s, whichever is larger;
- $\epsilon_{u4} = \pm 1^\circ C$;
- $\epsilon_{u4} = \pm 25, \pm 12, \pm 8, \pm 5$ and ±4% for velocities of 0.05–0.10, 0.10–0.15, 0.15–0.25, 0.25–0.51 and > 0.51 m/s, respectively;
- $\epsilon_{T} = \pm 0.4^\circ C$.

where, $\epsilon_{u4}$, $\epsilon_{T}$, $\epsilon_{u}$, and $\epsilon_{T}$ are uncertainties for diffuser air velocity, diffuser air temperature, room air velocities and room air temperatures, respectively. These uncertainties are important for interpreting the experimental results and for comparing measurements with numerical simulations.

Numerical Simulation Procedure

The computer code 'EXACT3' [10] was used for numerical simulation of the air and gas distribution within the office room. In 'EXACT3', the averaged three-dimensional Navier-Stokes equations coupled with the high Reynolds number k-ε turbulence model are solved numerically using the Marker and Cell finite difference method.

Beginning with the proper specifications of air conditions at the diffuser is critical to the success of numerical simulations of room airflow since they have the greatest influence on the room airflow pattern. Ideally, measured profiles of air velocity, temperature and turbulent kinetic energy at the diffuser are preferred for numerical simulations that are compatible with the experiments [5]. However, in most realistic cases, it is practically difficult (if not impossible) to measure the profiles of air velocity, temperature and turbulent kinetic energy, and these data are usually not available during the design stage. Therefore, certain types of approximations are usually necessary in specifying the boundary conditions at the diffuser.

For the simulation results presented in this paper, uniform profiles of air velocity, temperature and turbulent kinetic energy were assumed at the diffuser, but the magnitude of the diffuser air velocity was specified to provide an equal amount of jet momentum measured in the fully developed turbulent region of the diffuser air jets. Experiments showed that the jet momentum had a more significant effect on the room air motion than did the airflow rate [8]. Specifying diffuser air velocity based on the jet momentum has also been found to be important in previous numerical simulations [11].

The turbulent kinetic energy and temperature at the diffuser were specified based on the measurement at the centre of the diffuser slot, while the dissipation rate of turbulent kinetic energy was estimated by the following equation:

$$\varepsilon = C_p\frac{0.75k^{1.5}}{\nu}$$

where $\varepsilon$ is dissipation rate of turbulent kinetic energy, $C_p = 0.09$, $k$ is turbulent kinetic energy and $\nu = 0.03$ times the hydraulic diameter of the opening slot.

Surfaces of the ceiling and walls were assumed to be adiabatic. A uniform heat flux was specified for the floor surface.

The simulation results presented in this paper were obtained with a 50 × 44 × 23 non-uniform grid scheme. Simulation with a finer grid scheme (60 × 50 × 23) did not produce a different result in terms of room airflow patterns, so the 50 × 44 × 23 grid density was considered to be appropriate.

Simulations were conducted with a workstation computer which has 28.5 MIPS and 4.2 MFLOPS of integer and floating point performance, respectively. It took about 7 days CPU time to obtain the steady-state solution. The solution was considered to be steady state based on the following criteria: (1) the distribution patterns of air velocity and temperature remained unchanged; (2) the volumetric room mean square residuals of the equations of momentum, temperature, turbulent kinetic energy and dissipation rate of turbulent kinetic energy were less than $5 \times 10^{-5}$.

![Fig. 1. Schematic of the test room. 1 = Book shelves; 2 = desks; 3 = partitions.](image-url)
Results

Airflow Patterns
The observed airflow pattern showed an inclined diffuser air jet that attached to the ceiling due to the well-known Coanda effect and remained attached to the ceiling for a certain distance before it separated from the ceiling (fig. 2a). A reverse flow was formed under the diffuser air jet due to the entrainment of the jet. It can be seen that part of the jet fell to the occupied zone (defined as the region from the floor to 1.83 m and 0.30 m from each wall before it reached the opposite wall.

The computed airflow pattern (fig. 2b) agreed generally well with the observed pattern, but the jet did not drop until it reached the opposite wall, and the entire occupied zone was ventilated by the reverse flow. The computed jet also had a narrower spread than that observed with the helium bubbles.

Air Velocity Distribution
The contour map of the measured velocity distribution shows a decayed air jet, as expected (fig. 3a). If the 0.25 m/s contour line is regarded as the jet boundary, one can see the jet spread to the occupied zone before it reached the opposite wall, which was in agreement with the airflow pattern observed. Velocities in the occupied zone were fairly uniform (0.10–0.25 m/s) with low values at the centre of the large recirculation eddy (fig. 2a), as expected.

The computed velocity distribution pattern (fig. 3b) was similar to the measurement in general. That is, it predicted high velocities in the regions close to the ceiling, walls and floor and low velocities in the central region of the room. However, the computed contour map shows a slower decay of jet velocity, which results in higher velocities in the region close to the ceiling, walls and floor as compared with the measurements. The numerical model did predict low-velocity levels (0.10–0.20 m/s) at the central region of the room, which were similar to the measured values.
Fig. 4. Comparison between measured and computed contour maps of turbulent kinetic energy \([100 \times (m/s)^2]\) at the symmetric plane.

*Turbulent Kinetic Energy Distribution*

Measurements showed high turbulent kinetic energy in the jet region and low values in the occupied region, as expected (fig. 4a). There was relatively high turbulent kinetic energy close to the floor surface as compared with that at the centre of the occupied zone, which was due to the heat generation and the relatively higher velocity gradient over the surface.

The computed turbulent kinetic energy was significantly higher than the measurements (fig. 4b). This was due to the underestimation of the jet decay in the numerical model, since a higher velocity in the jet would cause a higher velocity gradient and cause more turbulence. The higher air velocity over the surfaces of the opposite wall and the floor would also cause more turbulence to be produced over these surfaces as compared with that in the experiment.

*Spatial Distribution of Temperature*

The contour map (fig. 5a) of the measured temperatures indicates that the incoming air temperature increased as the air travelled and mixed with the room air. Relatively high temperatures were present close to the floor surface due to the heat production there.

The numerical simulation predicted similar temperature distributions to the measured pattern (fig. 5b), but the predicted temperatures were slightly lower than the measurements in the occupied region.

*Discussion*

According to the above results, the numerical predictions are in reasonable agreement with the experimental measurements in terms of airflow pattern and the distribution patterns of air velocity, temperature and turbulent kinetic energy. However, the numerical simulation predicted slower jet decay, narrower jet spread and a delayed drop of the jet as compared with the measurements. As a result, the simulation overestimated the air velocity in the regions adjacent to the surfaces of the ceiling, walls and floor, and the turbulent kinetic energy in the room. The temperature in the room was underestimated.
These differences between the prediction and the measurement appear to indicate that the thermal buoyancy effect was not sufficiently accounted for in the numerical model, since the thermal buoyancy would speed up the jet decay, increase the jet spread and cause an early drop of the jet. However, the differences may be also due to the small three-dimensional effect present in the experiment. There can be up to 3% nonuniformity in the distribution of diffuser air velocity along the length of the diffuser slot (i.e., in direction z) and in the heat generation on the floor. This non-uniformity would cause air motion in the third direction (direction z). Such a three-dimensional effect would reduce the Coanda effect and result in an earlier drop of the jet. There were also uncertainties involved in the measurements, as described earlier, which might have added to the differences between the numerical and experimental results. For example, the single hot-wire probe used in the present measurement was not sensitive to the velocity in the third direction, which would cause error in the velocity measurement if there were a velocity component in that direction. ASHRAE [12] is currently sponsoring a research project to measure three-dimensional room air velocities for validating numerical simulation models.

The results showed that the predicted temperatures in the occupied region were lower than the measurements. This may be partially due to the radiation effect which was not modeled in the simulation code. Radiant heat exchange within the room, which obviously existed in the experiment, would make the room air temperature more uniform and result in a higher temperature in the occupied region when compared with the numerical simulation.

The differences between the simulation and measurement were also partially due to the approximation in specifying the boundary conditions in the simulation. In the practical application of numerical simulation, approximations in specifying the boundary conditions are unavoidable. Studies are needed to quantify the uncertainty involved in the simulation results due to this approximation in the boundary conditions.

Air distribution in a real room is a very complicated process. A systematic evaluation of the existing numerical simulation models should start with the simplest case (simple geometry, no internal heat load and no internal obstruction), and then gradually added complexity (internal heat load, obstructions and more complicated geometry). In this way, one can determine when the numerical models start to break down. Guidelines for the application of the models can be established once the limitations of the models are identified.

Conclusions

Systematic evaluation of numerical models of room air motion should be conducted from simple to complicated cases so that the limitations of the models can be clearly identified. Guidelines for using the numerical simulation technique can then be developed. For engineering design purposes, essential are the quantification of the uncertainties that may be involved in the specification of the boundary conditions of numerical simulations and the assessment of how such uncertainties affect simulation results. Criteria are also needed to judge what prediction accuracy is sufficient for engineering design purposes.

Nomenclature

\[ \begin{align*}
A_{\text{Ar}} &= \text{Archimedes number defined as } \frac{\beta g w_d (T_r - T_d)}{U_d^2}, \\
E(f) &= \text{spectral density function of velocity fluctuations, (m/s)}^2; \\
f &= \text{frequency, Hz}; \\
g &= \text{gravitational acceleration rate, m/s}^2; \\
H &= \text{room height, m}; \\
h &= \text{convective heat transfer coefficient, W/m}^2 \cdot \text{K}; \\
k &= \text{turbulent kinetic energy, (m/s)}^2; \\
L &= \text{length of the room (in direction z), m}; \\
L_d &= \text{length of the diffuser slot (in direction z), m}; \\
Q &= \text{ventilation rate, m}^3/\text{sec}; \\
q'' &= \text{heat flux from the floor surface, W/m}^2; \\
Re &= \text{Reynolds number defined as } \frac{U_d d}{v}, \\
T_r &= \text{maximum temperature in room (e.g., on the heated surface), °C}; \\
T_d &= \text{diffuser air temperature, °C}; \\
T_e &= \text{air temperature at the exhaust, °C}; \\
\Delta T_{\text{ed}} &= \text{jet momentum, m/s}; \\
u &= \text{standard deviation of velocity, m/s}; \\
U_o &= \text{nominal average air velocity at the diffuser calculated from the measured ventilation rate, m/s}; \\
U_d &= \text{average diffuser air velocity calculated from the measured jet momentum, m/s}; \\
U_e &= \text{average air velocity at the exhaust based on the mass balance, m/s}; \\
W &= \text{width of the test room (in direction x), m}; \\
W_d &= \text{width of the diffuser slot (in direction y), m}; \\
W_e &= \text{slot width of the exhaust, m}; \\
x, y, z &= \text{Eulerian Cartesian coordinates, m}; \\
y_d &= \text{y coordinate of the diffuser left edge, m}; \\
y_e &= \text{y coordinate of the exhaust right edge, m}; \\
\beta &= \text{thermal expansion coefficient, } 1/\text{K}; \\
\nu &= \text{kinematic viscosity, m}^2/\text{s}; \\
\rho &= \text{density of diffuser air, kg/m}^3; \text{and} \\
s_e &= \text{dissipation rate of the turbulent kinetic energy at the diffuser, (m/s)}^3/\text{s}. 
\end{align*} \]
References


