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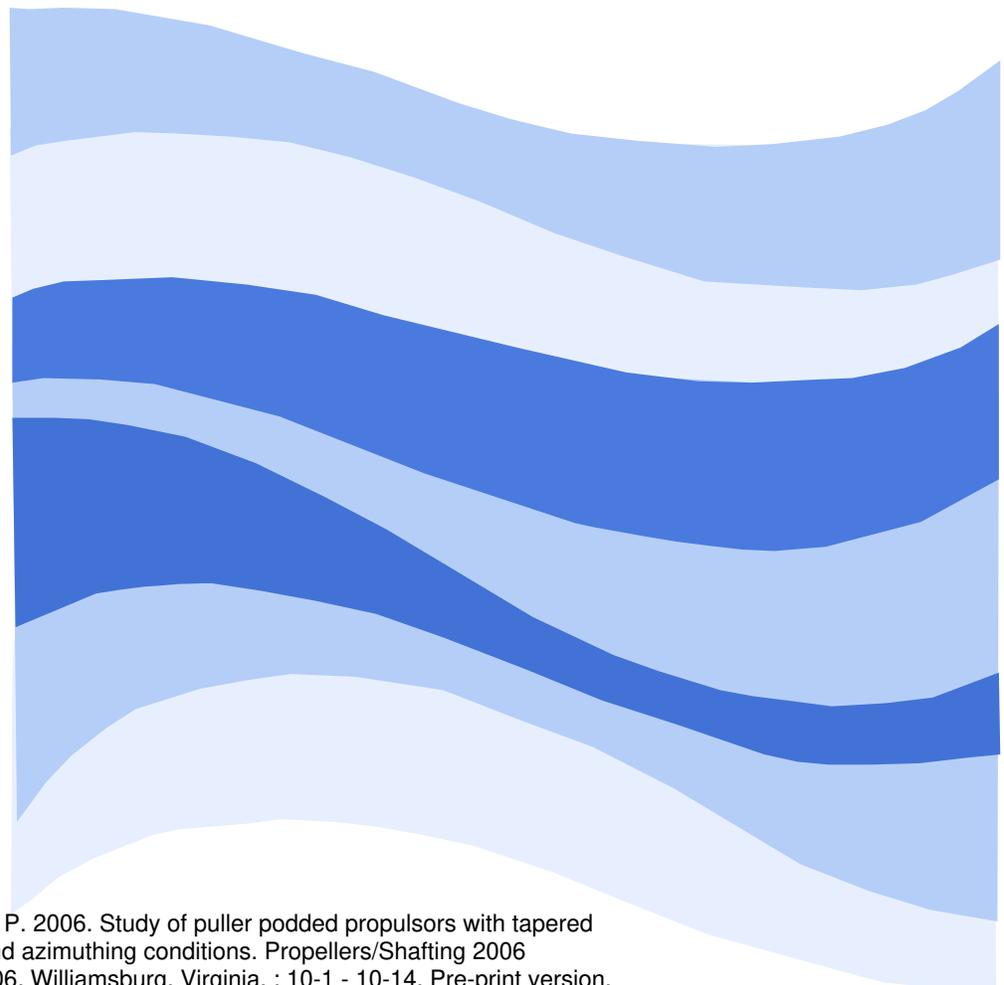
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## Study of puller podded propulsors with tapered hub propellers in straight course and azimuthing conditions

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# Hydrodynamic Characteristics of Puller Podded Propulsors with Tapered Hub Propellers

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## ABSTRACT

*This paper presents the results of an experimental study on the effects of propeller hub angle and azimuth conditions on the propulsive characteristics of puller podded propulsors in open water conditions. The propulsive performance of two model puller-podded units with different hub geometry was measured using a custom designed pod dynamometer in a towing tank. The podded units were tested to measure the thrust and torque of the propeller and the forces on the whole unit in three orthogonal directions. The tests were conducted for a range of advance coefficients of 0 to 1.2, combined with the range of static azimuth angles from  $+20^\circ$  to  $-20^\circ$  with a  $10^\circ$  increment. The dynamometer system consisted of a six-component global dynamometer and a three-component pod dynamometer. The variation in the propulsive performance because of the changes in hub geometry was examined first, followed by a study on the effect of azimuth conditions on the forces and moments on the propulsor. The results of the measurements are presented as the changes of forces and moments of the podded unit on the basis of advance coefficients and azimuth angles. The results illustrated that for the range of azimuth angles tested, the thrust and torque of the propeller and the forces on the whole unit in three orthogonal directions are complex functions of the azimuth angles for puller propulsors.*

## 1. INTRODUCTION

Podded propulsors have become a popular main propulsion system for ships due to their better hydrodynamic characteristics than the conventional propeller-rudder system and the advantages of diesel-electric propulsion. This propulsion unit is considered as a very useful combination of propulsion and steering system. It is accepted that a podded propulsor allows more flexibility in design of the internal arrangement of a ship, potentially reduced noise and vibration, and increased manoeuvrability, especially in a confined space Pakaste *et al.* (1999).

A research program on podded propellers is being undertaken jointly by the Ocean Engineering Research Centre (OERC) at Memorial University of Newfoundland (MUN), the National Research Council's Institute for Ocean Technology (IOT), Oceanic Consulting Corporation, and Thordon Bearings Ltd. The program combines parallel developments in numerical prediction methods and experimental evaluation. Amongst the hydrodynamic issues that have been identified are questions regarding the effects of hub taper angle (Islam (2004), Islam *et al.* (2004), Islam *et al.* (2005), Islam *et al.* (2006), Taylor *et al.* (2005), Taylor (2005)), pod-strut configuration (Islam (2004), Taylor (2005)), pod-strut interactions (He *et al.* (2005) and He *et al.* (2005)), gap pressure (MacNeill *et al.* 2004), and pod-strut geometry (Molloy *et al.* (2005) and Islam (2004)) on podded propeller performance. The present study focuses on the effect of propeller hub geometry and azimuth conditions of puller-podded propulsor performance.

Szantyr (2001a and 2001b) published one of the first sets of systematic experimental data on podded propulsors as the main propulsion unit with static azimuth angles. The tests measured the axial and transverse loads and used traditional non-dimensional coefficients to analyze the data. The study was limited to  $\pm 15^\circ$  azimuth angles. In the work, the effect of an azimuth angle on propeller torque was not studied. Grygorowicz and Szantyr (2004) presented open-water measurements of podded propulsors both in puller and pusher configurations in a circulating water channel. Heinke (2004) reported systematic model test results with a 4- and 5-bladed propeller fitted to a generic pod housing in pull- and push-mode. In the report, Heinke presented systematic data for forces and moments on the propeller and pod body at different static azimuth angles. Stettler *et al.* (2004) also investigated the dynamics of azimuth podded propulsor forces with emphasis on the application of nonlinear vehicle manoeuvring dynamics. There still exists considerable deficiency in the understanding of the hydrodynamic behavior of propellers and pods under different azimuth operating conditions.

The study on the hub geometry and azimuth conditions was aimed to help the better understanding of the behavior of forces and moments that act on the pods, which is important to design the propulsor. For instance, the study would quantify the relationship between azimuth conditions and bearing loads. In a study of podded propulsor failures, bearing failure was identified as one of the most significant cause of failure of the propulsors, Carlton (2002). This study would help the bearing designer to design pod bearings based on the loads that act on the propulsor at different azimuth conditions. Section 2 details the geometry of the propeller and pod-strut models used in this study, while section 3 presents a brief description of the apparatus and testing techniques used. Experimental results and discussions are provided in section 4, followed by summarizing conclusions in section 5.

## 2. EXPERIMENTAL SET-UP

### 2-1. Pod Models

The experiments included tests on two model propellers with two pods. The two propellers had identical blade section geometry but different hub taper angles of  $15^\circ$  and  $20^\circ$  (namely, Pull- $15^\circ$  and Pull- $20^\circ$ , respectively). The propellers were four bladed with a diameter of 0.27m, pitch-diameter ration (P/D) of 1.0 and expand area ration (EAR) of 0.6. The geometric particulars of the propellers are given in Islam (2004). The taper angles of the conical hubs were varied to study the influence of the taper angle on the performance of the propulsors.

The geometric particulars of the pod-strut models were defined using the parameters depicted in Figure 1. The values for the model propulsors were selected to provide an average representation of in-service, full-scale single screw podded propulsors. The particulars of the two pod-strut bodies tested are shown in Table 1. As the propeller hub taper angle was the only factor being considered in the first part of the study, only the fore taper angle was varied in the two pod-strut bodies. This was done to ensure a smooth flow transition between the propeller and the pod-strut body. Since all remaining parameters were held constant, Pod 1 with a  $15^\circ$  fore taper angle was used in combination with the Pull- $15^\circ$  propeller, and Pod 2 with a  $20^\circ$  fore taper angle was used in combination with the Pull- $20^\circ$  Propeller. For the study of performance in azimuth conditions, only Pod 2 was tested.

Table 1. Geometric particulars of the two pod-strut models.

External Dimensions of Model Pods	Pod 1 mm	Pod 2 mm
Propeller Diameter, $D_{Prop}$	270	270
Pod Diameter, $D_{Pod}$	139	139
Pod Length, $L_{Pod}$	410	410
Strut Height, $S_{Height}$	300	300
Strut Chord Length	225	225
Strut Distance, $S_{Dist}$	100	100
Strut Width	60	60
Fore Taper Length	85	85
Fore Taper Angle	15°	20°
Aft Taper Length	110	110
Aft Taper Angle	25°	25°

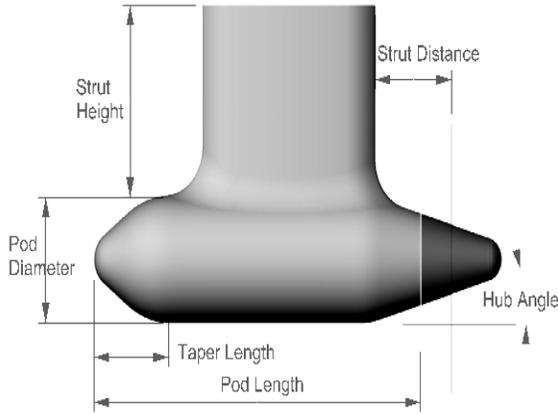
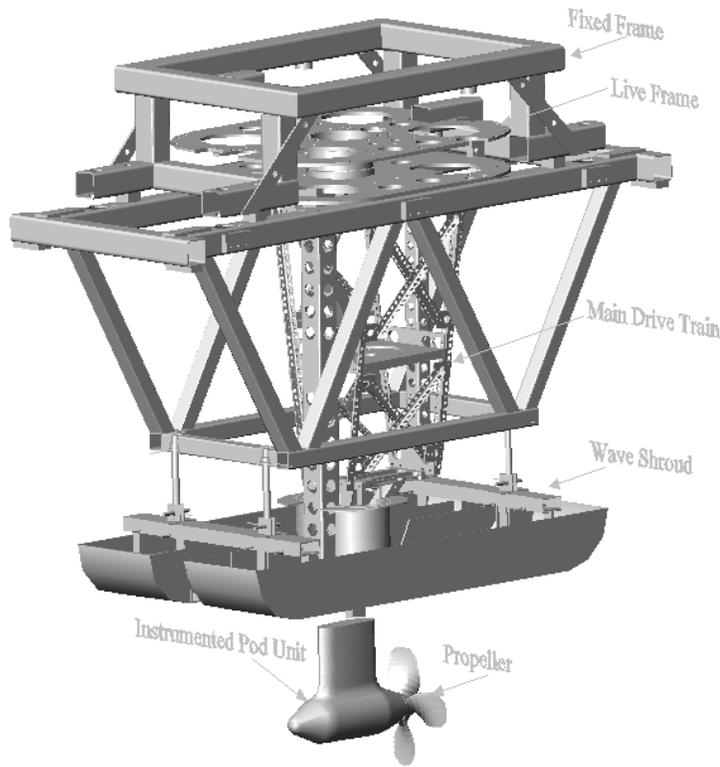


Fig. 1- Geometric parameters used to define pod-strut geometry.

## 2-2. Experimental Apparatus and Approach

The open water tests of the two pods in straight course and azimuth conditions were performed in accordance with the ITTC recommended procedure, Padded Propulsor Tests and Extrapolation, 7.5-02-03-01.3 (2002), and the description provided by Mewis (2001). A custom-designed dynamometer system called the NSERC-NRC pod dynamometer system (MacNeill *et al.* 2004) was used to measure propeller thrust, torque, and unit forces and moments. In the instrumentation, a motor fitted above the propeller boat drove the propeller via a belt system. The center of the propeller shaft was  $1.5D_{Prop}$  (propeller diameter) below the water surface. A boat shaped body called wave shroud was attached to the frame of the test equipment and placed just above the water surface. The boat stayed 3 to 5 mm above the water surface to avoid waves caused by the strut piercing the surface. The part of the shaft above the strut (the shaft connected the pod unit to the main drive of the equipment) went through the boat. Also, water temperature, carriage speed,  $V$ , and the rotational speed of the propeller,  $n$ , were measured. Figure 2 shows the different parts of the experimental apparatus.

As shown in Figure 2(a), the dynamometer system has two major parts. The first part is the pod dynamometer, which measures the thrust and torque of the propeller at the propeller shaft. The second part of the system is the global dynamometer, which measures the unit forces in three coordinate directions at a location above the propeller boat. A propeller boat was designed to minimize the surface wave effects. Further details of the experimental apparatus can be found in MacNeill *et al.* (2004). The propulsor was placed at different static azimuth conditions by rotating the entire lower part of the instrumentation (instrumented pod unit and the main drive as shown in Figure 2(a)). The entire lower part hung on a round plate, which had machined marks that defined the azimuth angles.



(a) NSERC-NRC pod dynamometer system.



(b) Propeller and the pod encasing the pod dynamometer



(c) Global dynamometer looking from below.



(e) Motor that runs the propeller with the gearbox.



(d) Motor used in the lifting system.

Fig. 2- Different parts of the experimental apparatus used in the pod series tests.

### 3. RESULTS AND DISCUSSIONS

The NSERC-NRC pod dynamometer system can measure propeller and pod forces and moments, namely: propeller thrust at hub end ( $T_{prop}$ ), propeller thrust at pod end ( $T_{pod}$ ), propeller torque ( $Q$ ), unit longitudinal force ( $F_x$ ) and moment ( $M_x$ ), unit transverse force ( $F_y$ ) and moment ( $M_y$ ), and unit vertical force ( $F_z$ ) and moment ( $M_z$ ).

For the study of hub taper angle, the measurements were done in straight course and in puller configurations using the two pods (Pod 1 and Pod 2).

For the study of azimuth conditions, the measurements were done in puller configurations using the Pod 2 at five different azimuth angles. The global dynamometer was calibrated using the method as described in David *et al.* (1980) and Galway (2000). The methods take into account cross talk between the six load cells and produced an interaction matrix to convert the voltage output into relevant forces and moments. The definition of the forces, moments and co-ordinates that were used to analyze the data and present the results is shown in Figure 3. The coordinate centre coincided with the



shown in the figure provides a comparison of the open water results for the two pods. Figure 4 shows the thrust and torque coefficients and propulsive efficiencies of only the propellers, propeller with 15° taper angle and propeller with 20° taper angle, in the two pod units, Pod 1 and Pod 2, respectively. Figure 5 shows the thrust coefficients and propulsive efficiencies of the two pod units (pod-strut-propeller as a whole unit) Pod 1 and Pod 2. All the measurements were conducted at a constant 660 rpm both for the positive direction of rotation, Pull-20° (i.e. clockwise rotation viewed from downstream) and negative direction of rotation, Pull-15° (i.e. anticlockwise rotation viewed from downstream). The analysis of the repeated tests showed that the repeatability for the thrust and torque coefficients stayed within 2%. Taylor (2005) presented an uncertainly analysis of the test equipment.

To assess the influence of the hub taper angle on the podded propeller units, the differences in performance coefficients of the two pods are shown in Table 3. The percent values were calculated using equation 1.

$$\%Diff = (X_{pod2} - X_{pod1}) \times 100 / X_{pod2|J=0.0} \dots (1)$$

From Figures 4 and 5, and Table 3, it is observed that for two podded units in puller configurations, increasing the hub taper angle had noticeable effect on propeller thrust and torque coefficient and efficiency in

the range of advance coefficients tested. Increasing hub taper angle had an increasing effect on thrust coefficient as the advance coefficient was increased from 0.0 to 0.5. However, the thrust was decreased for the propeller with increasing hub angle as the advance coefficient was above 0.6. The Pull-20° propeller produced 2% more and 01% less thrust than the Pull-15° propeller at advance coefficient of 0.5 and 0.8, respectively. The propeller shaft torque coefficient was increased by a small amount (within 0.3%) in the range of advance coefficient of 0.0 to 0.6, as the hub angle was increased. The propeller shaft torque coefficient decreased approximately 2.0% as the hub angle was increased (when the advance coefficients were above 0.6). This resulted in noticeable increase in propeller efficiency with an increase in hub angle (4.0% increase in  $\eta_{prop}$  at  $J=0.8$ ). However, Pull-20° produced higher unit thrust at all advance coefficients, and the maximum increase was seen at moderate advance coefficients of  $J=0.5\sim0.8$  (approximately 5% increase at  $J=0.8$ ). Similarly to the propeller efficiency or the unit thrust coefficient, the pod unit with a higher hub angle provided higher unit efficiency. Also, the maximum increase of unit efficiency for the propeller with a higher hub angle was seen at the advance coefficient of  $J=0.8$  (approximately 13.0%).

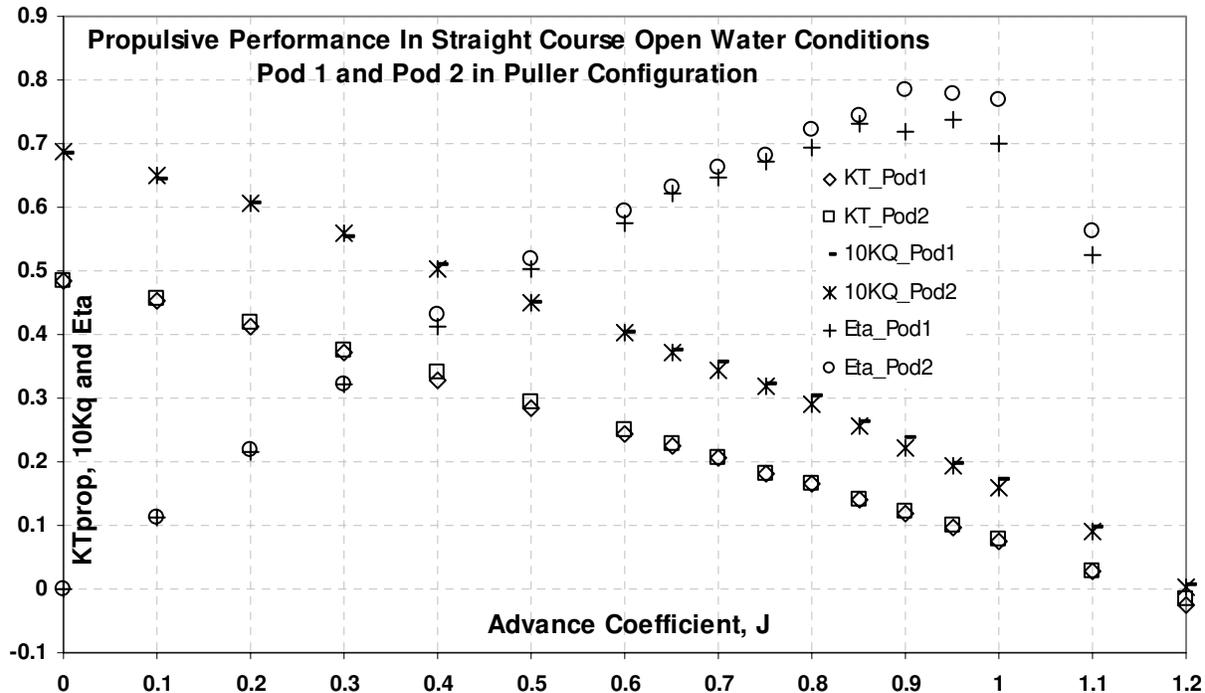


Fig. 4- Propulsive performance of the propeller “only” for the propulsors, Pod 1 (propeller with 15° hub) and Pod 2 (propeller with 20° hub).

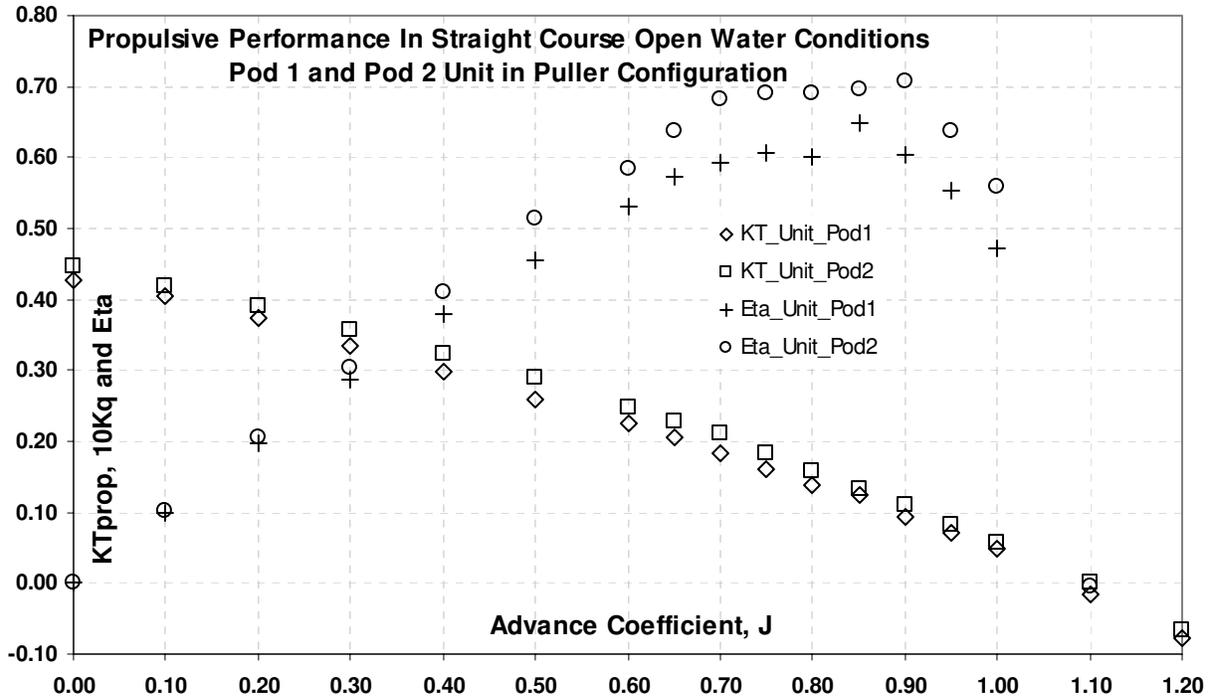


Fig. 5- Propulsive performance for the whole pod units (pod-strut-propeller): Pod 1 and Pod 2.

Table 3. Percent Difference between 20° Pull Podded Propeller and 15° Pull Podded Propeller Results

J	% Diff $K_Q$	% Diff $K_{Tprop}$	% Diff $\eta_{prop}$	% Diff KT_unit	% Diff $\eta_{unit}$
0.0	0.3	-0.1	0.0	4.5	0.0
0.2	0.0	1.1	0.3	3.8	1.2
0.5	0.2	1.9	1.9	7.3	8.2
0.7	-1.6	-0.2	2.3	6.2	12.6
0.8	-1.8	-0.1	3.7	4.5	12.5
0.9	-1.9	0.7	8.1	3.6	14.6

### 3-2. Influence of Azimuth Conditions on Propeller Forces

The coefficients of the propulsor in puller configurations at different azimuth conditions are influenced by the hub geometry, propeller rotation direction and the interaction with pod-strut housing. The interaction effect is essentially because of the heterogeneous flow distribution in the propeller plane, induced by the propeller blades, strut and pod housing. The thrust and torque coefficients and propulsive efficiency of “only” the propeller, obtained from the tests in open water azimuth conditions (with the Pod 2 in puller configuration) are shown in figures 6 and 7. The propeller thrust was in the direction of propeller shaft (see Figure 3). The tests were done in five static azimuth angles ranging from  $-20^\circ$  to  $20^\circ$  in an increment of  $10^\circ$ . The positive direction of azimuth angle for the pod unit is shown in figure 3. In figures 6 and 7, 10Port means the pod unit was placed at  $10^\circ$

away from the straight course position in an anticlockwise direction (looking toward the pod unit from top).

Figure 6 shows that the propeller thrust and torque coefficients and propulsive performance remained approximately the same for  $20^\circ$  (Port) and  $-20^\circ$  (Starboard) static azimuth angles (for all the advance coefficients tested). The same conclusion applies for  $10^\circ$  (Port) and  $-10^\circ$  (Starboard) static azimuth angular positions. The propeller efficiency at straight-ahead condition was lowest and at  $20^\circ$  Port/Starboard condition was highest. At advance coefficient,  $J=1.0$ , the propeller efficiency was maximum when the pod unit was at  $20^\circ$  azimuth angle position in port or starboard side. It should be noted that Figure 6 and 7 shows results only for the propeller only not the whole unit. Figure 7 (a) and (b) show the variation of propeller thrust and torque with the azimuthing angles at a fixed

advance coefficients as indicated in the legends. The Figures show that propeller thrust and torque were minimum when it was operating at straight course condition, compared to  $\pm 10^\circ$  and  $\pm 20^\circ$  azimuth conditions (for the entire range of advance coefficient). Table 4 shows the performance variations of the propeller operating in straight course conditions compared to the azimuth conditions. In the calculation of differences in thrust and torque coefficients, equation 1 was used. In the calculation of differences in propulsive efficiency, equation 2 was used.

$$\%Diff\_Eta = \left( \frac{Eta_{straight-ahead} - Eta_{azimuth}}{Eta_{straight-ahead}|_{J=0.8}} \right) \dots(2)$$

As shown in table 4, and Figures 7(a) and 7(b), for all advance coefficients tested, the thrust and torque coefficients and propulsive efficiency for the propeller “only” were the lowest at straight course operating conditions (as indicated by negative sign of % Diff

values in table 4). At the design advance coefficient of  $J=0.8$ , an increase of 11.0% of  $K_{Tprop}$  was seen for the propulsors with an azimuth angle of  $20^\circ$  port side, as compared to that of the propulsor in straight course. In the same operating conditions, an increase of 7.0%, 6.0% and 12.0% was seen for the propulsors in  $10^\circ$ ,  $-10^\circ$  and  $-20^\circ$  azimuth angles. Thus, the percentage increases of thrust because of azimuth on either direction (Port or Star side) from the straight course condition are almost the same. Similar results were observed for all other advance coefficients. The percentage change of torque coefficients and propulsive efficiency of the propeller also behaved like that of the thrust coefficients. At advance coefficient of 0.8, an increase of 19.0%, 17.0%, 16.0% and 20.0% was seen for the propulsors in  $20^\circ$ ,  $10^\circ$ ,  $-10^\circ$  and  $-20^\circ$  azimuth angles. The results also showed that, as the azimuth angles changed from  $0^\circ$  to  $+20^\circ$  or from  $0^\circ$  to  $-20^\circ$ , the increases of thrust, torque and efficiency were nonlinear with the change of azimuth angles.

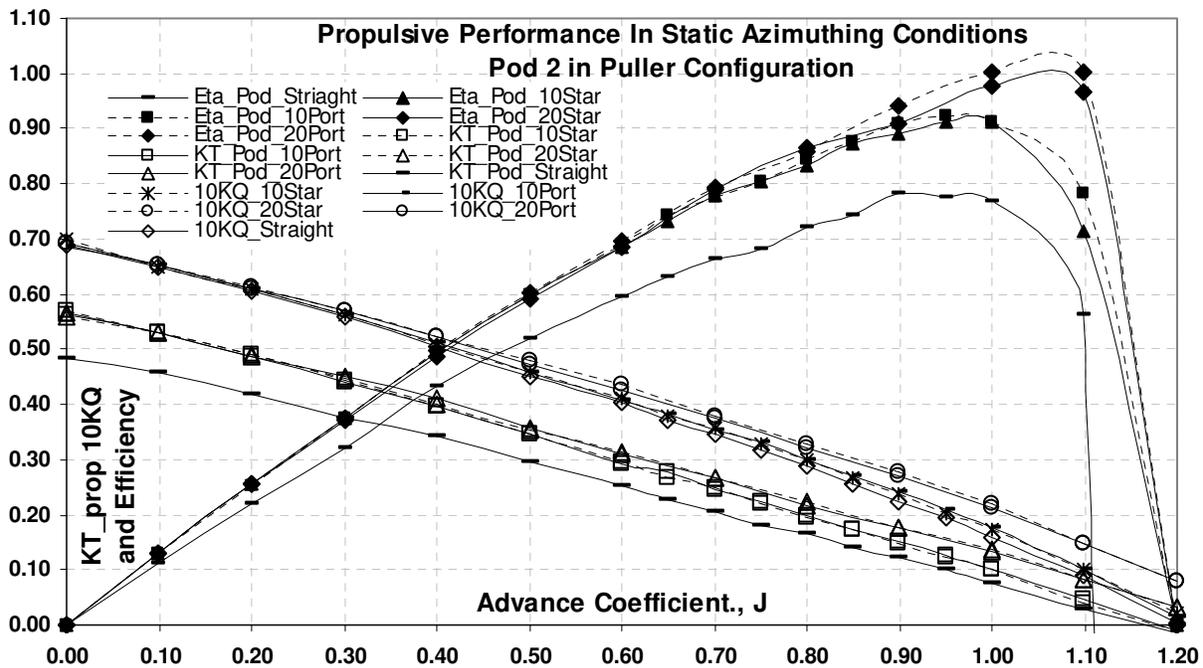


Fig. 6- propulsive performance plots for the Pod 2 unit at different azimuth conditions.

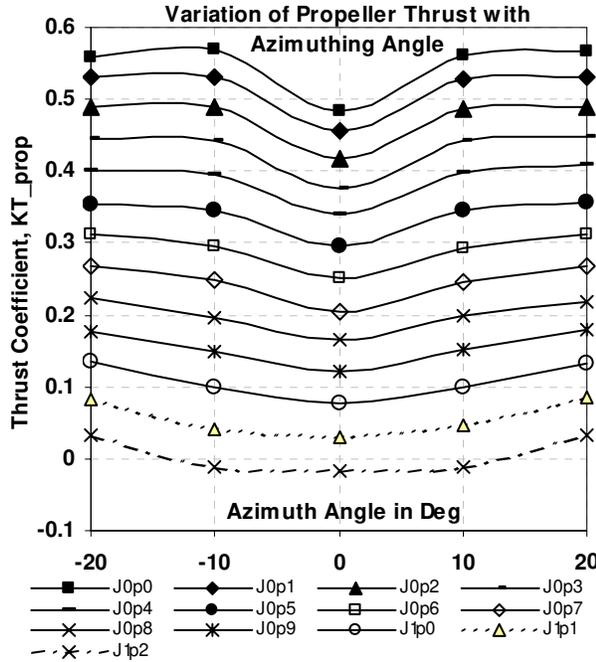


Fig. 7(a)- Variation of thrust coefficients with azimuth angles at fixed advance coefficients.

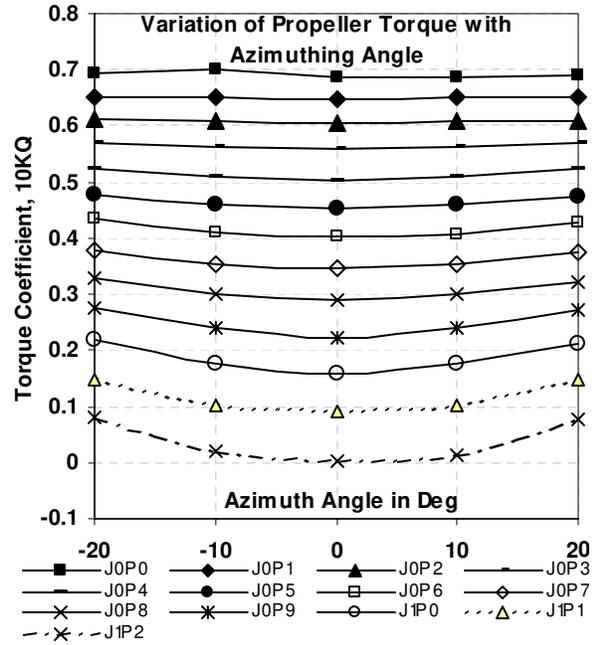


Fig. 7(b)- Variation of torque coefficients with azimuth angles at fixed advance coefficients.

Table 4. Percent Difference in 20° Pull Podded Propeller and 15° Pull Podded Propeller Results

J	% Diff in $K_{Tprop}$				% Diff in $10K_Q$				% Diff in $\eta_{prop}$			
	$K_{Tprop}$		-20°		$10K_Q$		$\eta_{prop}$		-20°		$\eta_{prop}$	
	20°	10°	-10°	Star	20°	10°	-10°	Star	20°	10°	-10°	Star
0.0	-17.1	-16.0	-18.0	-15.2	-0.6	0.1	-1.8	-0.8	0.0	0.0	0.0	0.0
0.2	-14.4	-13.9	-14.5	-14.9	-0.5	-0.4	-0.3	-0.9	-4.9	-4.7	-5.0	-4.9
0.4	-14.2	-11.6	-11.3	-12.2	-2.9	-1.1	-0.8	-2.8	-9.3	-8.8	-8.8	-7.7
0.6	-12.4	-8.7	-8.9	-12.6	-3.3	-0.5	-0.9	-4.4	-14.3	-13.0	-12.7	-12.8
0.8	-10.9	-7.2	-6.5	-12.3	-4.8	-1.5	-1.3	-5.8	-18.8	-16.9	-15.4	-19.9
1.0	-11.6	-4.7	-4.7	-12.0	-7.4	-2.2	-2.1	-8.8	-32.7	-19.9	-19.8	-29.0

### 3-3. Influence of Azimuth Conditions on Unit Forces

The forces and moment coefficients of Pod 2 for the range of advance coefficients and azimuth angle tested are presented in figures 8 to 12. The longitudinal force coefficients,  $K_{TX}$  (also called unit thrust,  $K_{Tunit}$ ) decreased as the advance coefficients increased (see Figure 8). As the azimuth angles were increased from 0° to 20° or from 0° to -20°, the  $K_{TX}$  decreased. An exception occurred when the azimuth angles were changed from 0° to 10° (Port side) where a small increase of  $K_{TX}$  was seen. The reduction of the longitudinal force was stronger for the negative azimuth direction, i.e. for right hand propeller, the clockwise

azimuth direction (in the present case, the -10° and -20° azimuth conditions, see Figure 13b).

Figures 9 and 13(c) show the change of transverse force coefficients with advance coefficient and azimuth angles (at different fixed  $J$ s). The propulsors with positive azimuth angles showed an increasing transverse force with the increase of  $J$ , and the propulsors with negative azimuth angles showed a decreasing transverse force with the increase of  $J$ . The zero transverse force was found in the range of azimuth angles from -1° to -3.5° (clockwise azimuth) for all of the advance coefficients. The vertical force coefficient,  $K_{FZ}$ , also showed a similar trend as that of the  $K_{FY}$  (see Figures 10 and 13(d)). It showed an increasing trend

with increase of azimuth angles and the zero vertical force occurred in the range of azimuth angles from  $0^\circ$  to  $14^\circ$  (counter-clockwise azimuth) for the range of advance coefficients tested. The moments around x- and y-axis ( $K_{MX}$  and  $K_{MY}$ ) are dependent on the longitudinal and transverse forces, and the propeller torque. These moments also showed an increasing trend with increase of azimuth angles (see Figures 11 and

13(e)). The steering moment (vertical moment about z-axis) became zero in the range of azimuth angles from  $4^\circ$  to  $10^\circ$  (counter-clockwise azimuth) for the range of advance coefficients tested. As the azimuth angle increased, the steering moment increased in a non-linear fashion for the range of  $-20^\circ$  to  $20^\circ$  azimuth angles (see Figures 12 and 13f).

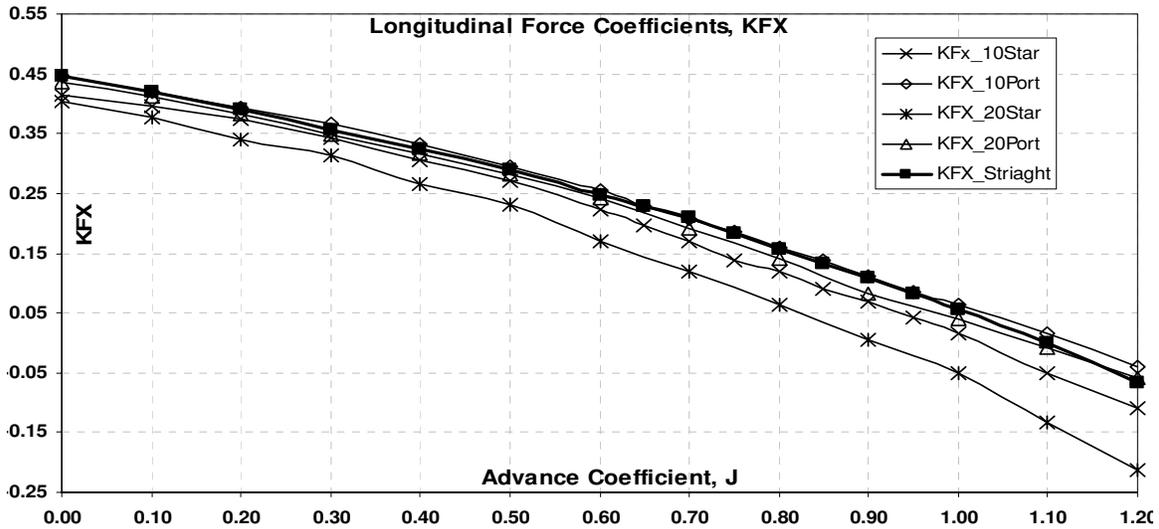


Fig. 8- longitudinal force coefficient plots for the Pod 2 at different azimuth conditions.

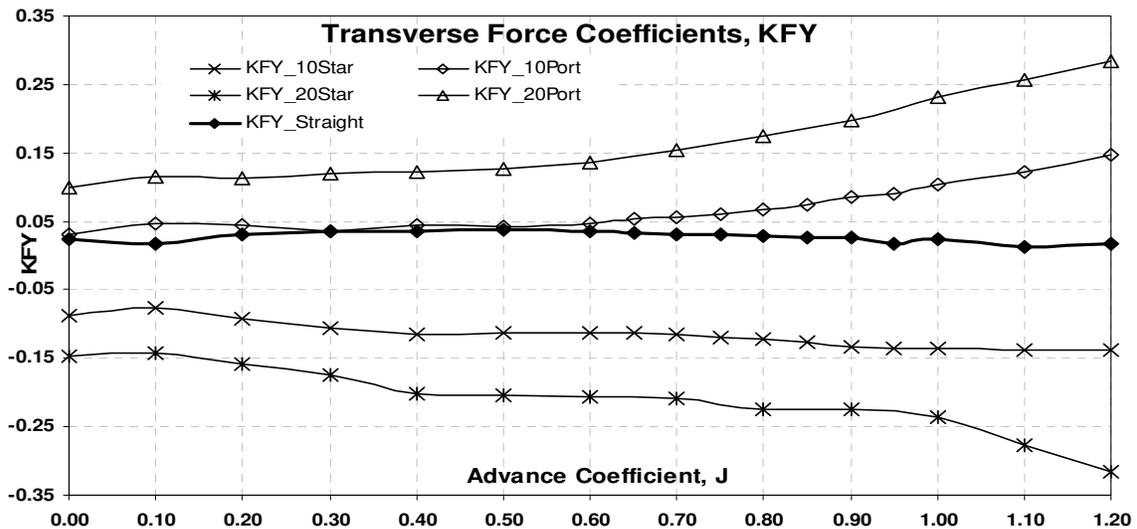


Fig. 9- Transverse force coefficient plots for the Pod 2 at different azimuth conditions.

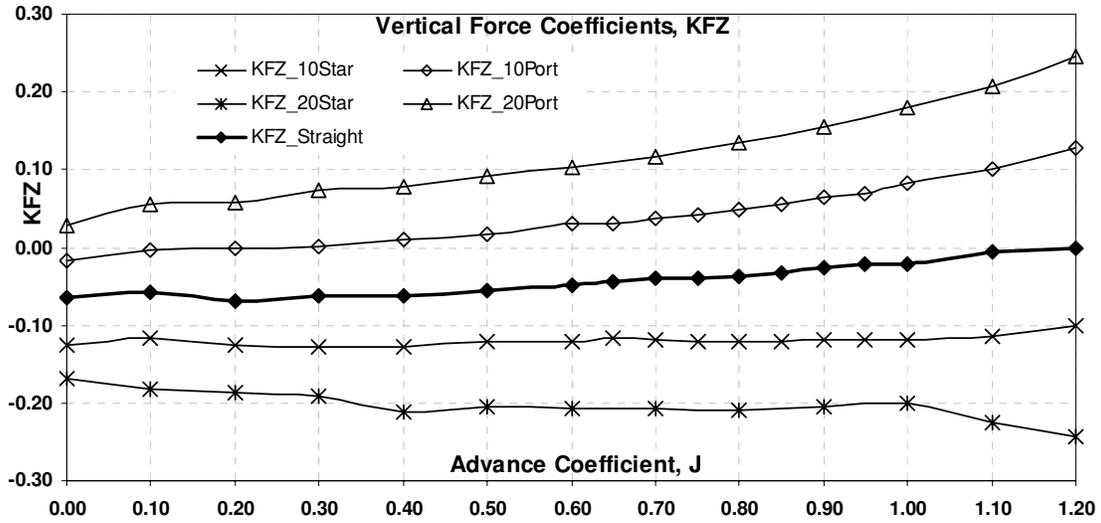


Fig. 10- Vertical force coefficient plots for the Pod 2 at different azimuth conditions.

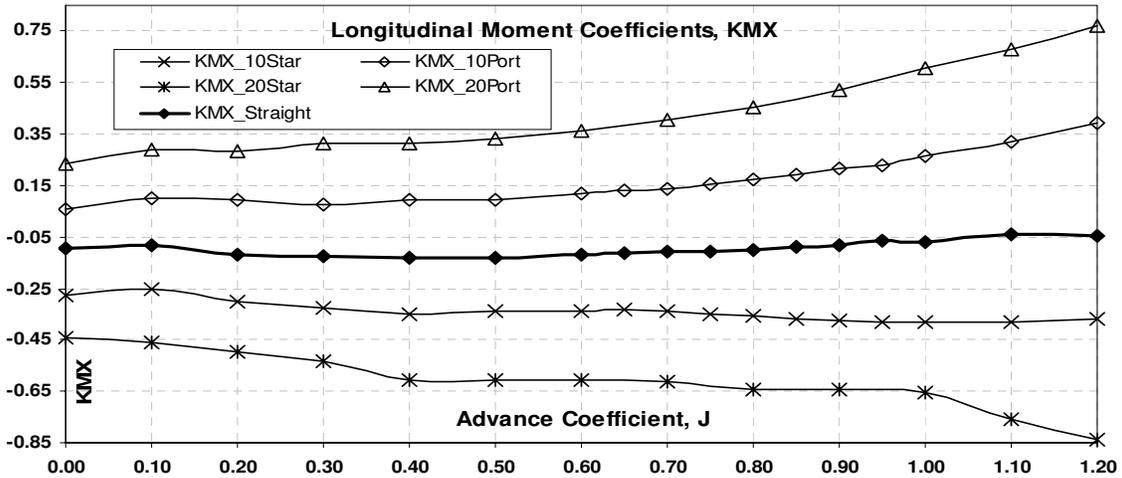


Fig. 11- longitudinal moment coefficient plots for the Pod 2 at different azimuth conditions.

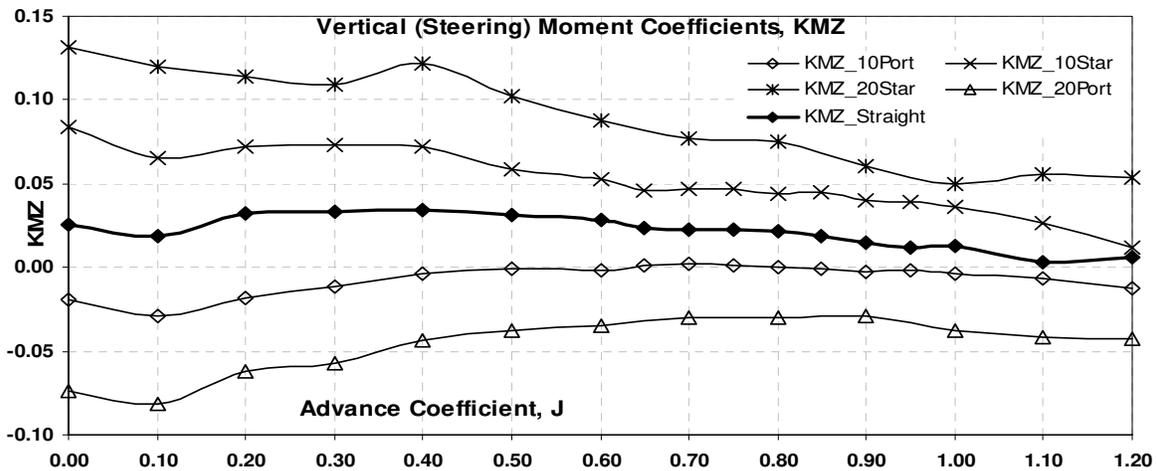
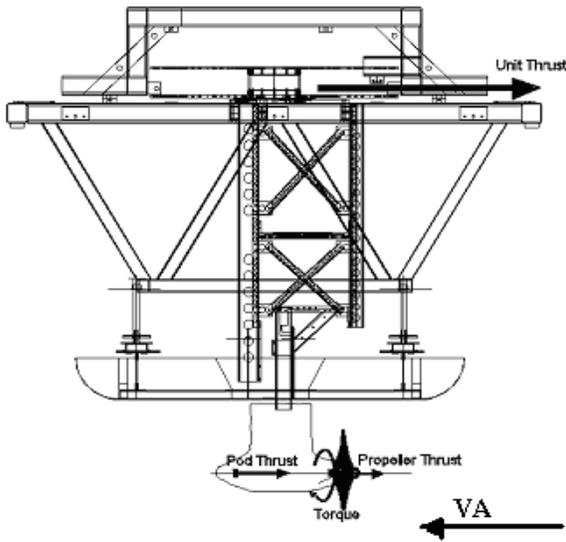
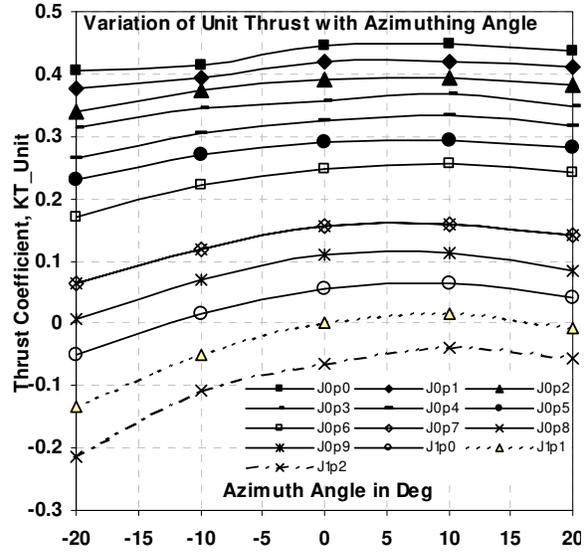


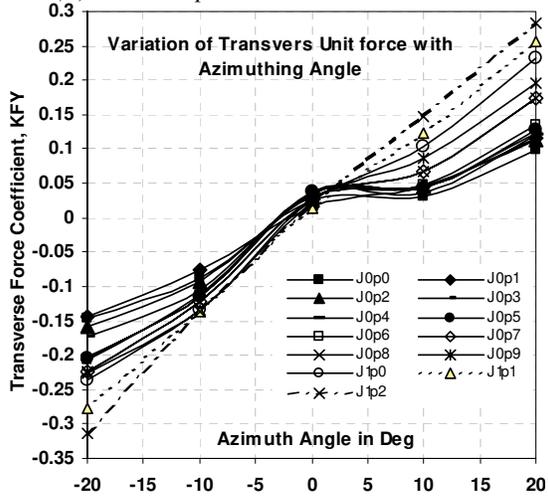
Fig. 12- Vertical (steering) moment coefficient plots for the Pod 2 at different azimuth conditions.



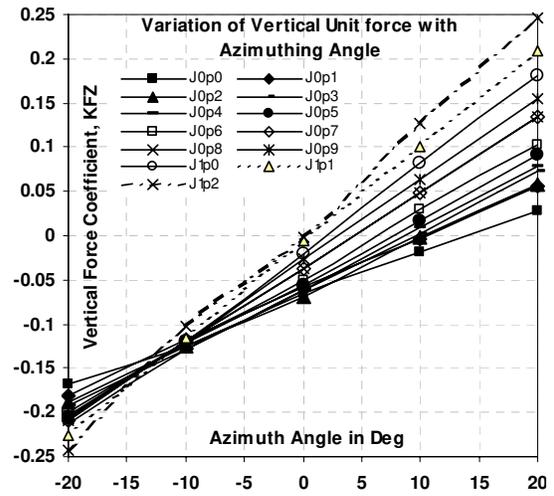
(a) Test set-up and measurement location



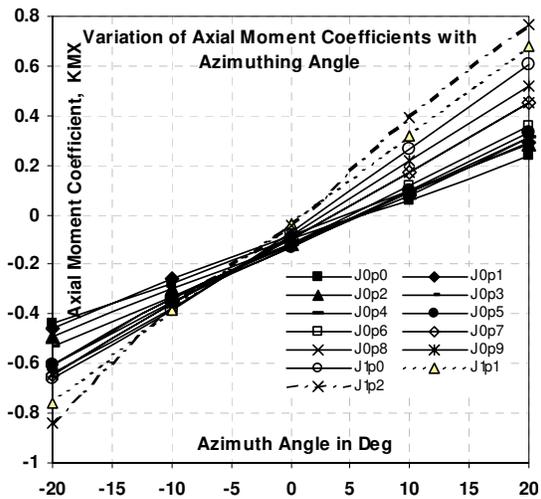
(b) Longitudinal force coefficient



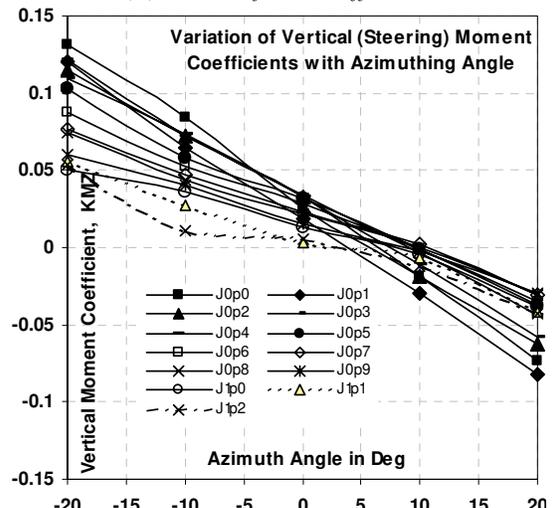
(c) Transverse force coefficient



(d) Vertical force coefficient



(e) Longitudinal moment coefficient



(f) Vertical moment coefficient

#### 4 CONCLUDING REMARKS

The present set of experiments investigated the effects of propeller hub taper angle and azimuth angle on the propulsive performance of podded propulsors in puller configuration. Two pods were tested using the pod testing system at the OERC towing tank at Memorial University. The results have provided valuable insight into the effects of hub taper angle and azimuth angle on the propulsive performance of podded propulsor.

The first set of experiments on the study of hub taper angle showed that, increasing the hub taper angle had an increasing effect on propeller efficiency in the range of advance coefficients tested. The propeller torque coefficient decreased approximately 2.0% as the hub angle was increased in the range of advance coefficients of 0.6 to 1.0. This resulted in a noticeable increase in propeller efficiency with increase in hub angle (4.0% increase in propeller efficiency at advance coefficient of 0.8). The podded unit with higher hub taper angle always produced higher unit thrust and the maximum increase was seen at moderate advance coefficients of 0.5 to 0.8 (approximately 5.0% increase at advance coefficient of 0.8). Similarly to the propeller efficiency and unit thrust coefficient, the pod unit with higher hub angle provided higher unit efficiency and maximum increase was seen at around design advance coefficient 0.8 (approximately 13.0% increase).

The influence of azimuth angle on the characteristics of puller-podded propulsors was dependent on the magnitude and direction of the azimuth angle. However, the thrust and torque of only the propeller were more or less independent of the azimuth direction. The thrust and torque coefficients for the propeller were lowest at straight course conditions. At the design advance coefficient of 0.8, an increase of 11.0% of propeller thrust coefficient was seen for the propulsors with an azimuth angle of 20°, as compared to that of the propulsor in straight course. In the same operating conditions, an increase of 7.0%, 6.0% and 12.0% was seen for the propulsors in 10°, -10° and -20° azimuth conditions. Similar results were observed for all other advance coefficients. The percentage change of torque coefficients and propulsive efficiency also behaved like that of the thrust coefficients. At advance coefficient of 0.8, an increase of 19.0%, 17.0%, 16.0% and 20.0% was seen for the propulsors in 20°, 10°, -10° and -20° azimuth angles. The results also showed that, as the azimuth angles changed from 0° to +20° or from 0° to -20°, the increases of thrust, torque and efficiency were nonlinear with the change of azimuth angles. It should be noted that, the increase of thrust, torque and efficiency of the propeller not necessarily means an increase in the whole unit thrust and efficiency, where the drag of the pod-strut is also included.

The force and moment coefficients of the propulsors showed a strong dependence on the propeller advance coefficient and azimuth angle. The longitudinal force coefficient was decreasing with the increasing advance coefficients and for both azimuth directions ( $\pm 20^\circ$ ). The reduction of the longitudinal force was stronger for the negative azimuth direction, i.e. for right hand propeller, the clockwise azimuth direction (in the present case, the  $-10^\circ$  and  $-20^\circ$  azimuth conditions). The propulsors with positive azimuth angles shows as increasing transverse force with the increase of  $J$  and vice versa. The zero transverse force was found in the range of azimuth angles from  $-1^\circ$  to  $-3.5^\circ$  (clockwise azimuth) for all of the advance coefficients. The vertical force coefficient also showed a similar trend as that of the transverse force coefficient. It showed an increasing trend with increase of azimuth angles and the zero vertical force occurred in the range of azimuth angles from  $0^\circ$  to  $14^\circ$  (counter-clockwise azimuth) for the range of advance coefficients tested. The moments around x- and y-axis also showed an increasing trend with increase of azimuth angles. The steering moment (vertical moment about z-axis) became zero in the range of azimuth angles from  $4^\circ$  to  $10^\circ$  (counter-clockwise azimuth) for the range of advance coefficients tested. As the azimuth angle increased, the steering moment increased in a non-linear fashion for the range of  $-20^\circ$  to  $20^\circ$  azimuth angles.

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